

above. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 95-CE-36-AD." The postcard will be date stamped and returned to the commenter.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

The FAA has determined that this regulation is an emergency regulation and that must be issued immediately to correct an unsafe condition in aircraft, and is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket (otherwise, an evaluation is not

required). A copy of it, if filed, may be obtained from the Rules Docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 USC 106(g), 40101, 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

95-17-07 **Fairchild Aircraft:** Amendment 39-9336; Docket No. 95-CE-36-AD.

Applicability: The following airplane models and serial numbers, certificated in any category:

Model	Serial Nos.
SA226-T	T201 through T75 and T77 through T291.
SA226-T(B)	T(B)276 and T(B)292 through T(B)417.
SA226-AT	AT001 through AT074.
SA226-TC	TC201 through TC419.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (d) of this AD to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition, or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or

repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated in the body of this AD.

To prevent the main landing gear (MLG) doors from jamming, which could prevent the gear from extending and could result in an emergency gear-up landing, accomplish the following:

(a) For airplanes with the MLG door stop that has been installed in accordance with Fairchild Aircraft SA226-32-043, Revised: September 6, 1983, within the next 50 hours time-in-service after the effective date of this AD, unless already accomplished, remove the part number (P/N) 27K51001-007 bolts (4 each) or alternate P/N NAS 428-3-13 bolts and jam nuts, P/N NAS 509-3 (4 each), from the P/N 27K51001-003 bracket assembly.

(b) For all affected airplanes, within the next 50 hours TIS after the effective date of this AD, unless already accomplished within the last 400 hours TIS, accomplish the "Fluid Servicing—Landing Gear" instructions in SA226 Maintenance Manual 32-00-00, paragraph 1C.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Airplane Certification Office (ACO), FAA, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Fort Worth ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Fort Worth ACO.

(e) All persons affected by this directive may obtain copies of Fairchild Aircraft Service Bulletin 226-32-043, Revised: September 6, 1983, or page 1C of SA226 Maintenance Manual 32-00-00 upon request to Fairchild Aircraft, P.O. Box 790490, San Antonio, Texas 78279-0490; or may examine this service bulletin at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

(f) This amendment (39-9336) becomes effective on August 28, 1995.

Issued in Kansas City, Missouri, on August 10, 1995.

Gerald W. Pierce,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-20278 Filed 8-18-95; 8:45 am]

BILLING CODE 4910-13-U

14 CFR Part 39

[Docket No. 95-CE-13-AD; Amendment 39-9339; AD 95-17-09]

Airworthiness Directives; Fairchild Aircraft SA226 and SA227 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Fairchild Aircraft SA226 and SA227 series airplanes that utilize a direct current (DC) generator. This action requires relocating the left-hand (LH) and right-hand (RH) essential bus current limiters (225 amp) to the battery bus (main bus tie). A safety recommendation received by the Federal Aviation Administration (FAA) that details potential electrical failure problems on Fairchild Aircraft SA226 and SA227 series airplanes prompted this action. The actions specified by this AD are intended to prevent failure of the LH and RH essential bus when engine failure results in a blown generator current limiter, which could result in loss of airplane electrical power.

DATES: Effective October 3, 1995.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of October 3, 1995.

ADDRESSES: Service information that applies to this AD may be obtained from Fairchild Aircraft, P.O. Box 790490, San Antonio, Texas 78279-0490; telephone (210) 824-9421. This information may also be examined at the FAA, Central Region, Office of the Assistant Chief Counsel, Attention: Rules Docket No. 95-CE-13-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Mr. George R. Hash, Aerospace Engineer, FAA, Airplane Certification Office, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150; telephone (817) 222-5134; facsimile (817) 222-5959.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Fairchild Aircraft SA226 and SA227 series airplanes that utilize a DC generator was published in the **Federal Register** on March 16, 1995 (60 FR 14235). The action proposed to require relocating the LH and RH essential bus current limiters (225 amp) to the battery

bus (main bus tie). Accomplishment of the proposed modification would be in accordance with Fairchild Aircraft Engineering Kit Drawing 27K82376, "Current Limiter Reusing Kit," as referenced in Fairchild Service Bulletin (SB) 226-24-034, SB 227-24-015, and SB CC7-24-002, all Issued: September 29, 1994.

Interested persons have been afforded an opportunity to participate in the making of this amendment. One comment was received in favor of the proposal and no comments were received regarding the FAA's estimate of the cost impact upon the public.

After careful review of all available information related to the subject presented above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. The FAA has determined that these minor corrections will not change the meaning of the AD and will not add any additional burden upon the public than was already proposed.

The FAA estimates that 622 airplanes in the U.S. registry will be affected by this AD, that it will take approximately 4 workhours per airplane to accomplish the required action, and that the average labor rate is approximately \$60 an hour. Parts cost approximately \$98 per airplane. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$210,236 or \$338 per airplane. This figure is based on the assumption that no affected airplane owner/operator has incorporated the required modification. Fairchild Aircraft has informed the FAA that parts have not been distributed to any owner/operator of the affected airplanes.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory

Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 USC 106(g), 40101, 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

95-17-09 Fairchild Aircraft: Amendment 39-9339; Docket No. 95-CE-13-AD.

Applicability: The following model and serial number airplanes that utilize a direct current (DC) generator, certificated in any category.

Models	Serial Nos.
SA226-T, SA226-AT, SA226-TC, and SA226-T(B).	All
SA227-AC, SA227-AT, SA227-BC, and SA227-TT.	1 through 733.
SA227-CC and SA227-DC.	784, and 790 through 883.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (c) of this AD to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition, or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required within the next 2,000 hours time-in-service after the effective date of this AD, unless already accomplished.

To prevent failure of the left hand (LH) and right hand (RH) essential bus when engine failure results in a blown generator current limiter, which could result in loss of airplane electrical power, accomplish the following:

(a) Relocate the LH and RH essential bus current limiters (225 amp) to the battery bus (main bus tie) in accordance with Fairchild Aircraft Engineering Kit Drawing 27K82376, "Current Limiter Reusing Kit," as referenced in the following service bulletins (SB):

SB	Date	Models affected
226-24-034	September 29, 1994.	All affected SA226 models.
227-24-015	September 29, 1994.	SA227-AD, SA227-AT, SA227-BC, and SA227-TT.
CC7-24-002	September 29, 1994.	SA227-CC and SA227-DC.

(b) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Fort Worth Airplane Certification Office (ACO), FAA, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Fort Worth ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Fort Worth ACO.

(d) The modification required by this AD shall be done in accordance with Fairchild Aircraft Engineering Kit Drawing 27K82376, "Current Limiter Reusing Kit," as referenced in Fairchild Aircraft Service Bulletins 226-24-034, 227-24-015, and CC7-24-002, all Issued: September 29, 1994. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Fairchild Aircraft, P.O. Box 790490, San Antonio, Texas 78279-0490. Copies may be inspected at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW., 7th Floor, suite 700, Washington, DC.

(e) This amendment (39-9339) becomes effective on October 3, 1995.

Issued in Kansas City, Missouri, on August 10, 1995.

Gerald W. Pierce,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-20275 Filed 8-18-95; 8:45 am]

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14 CFR Part 39

[Docket No. 95-NM-125-AD; Amendment 39-9316; AD 95-15-11]

Airworthiness Directives; Jetstream Model 4101 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that is applicable to certain Jetstream Model 4101 airplanes. This action requires a visual inspection of the passenger seat belt assemblies to determine if certain seat belt shackles are installed, and replacement of those shackles with improved ones. This amendment is prompted by reports indicating that certain passenger seat belt shackles have failed. The actions specified in this AD are intended to prevent reduced effectiveness of the passenger safety restraint system during take-off and landing or in an emergency situation, due to failure of the passenger seat belt shackles.

DATES: Effective September 5, 1995.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of October 20, 1995.

Comments for inclusion in the Rules Docket must be received on or before October 20, 1995.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 95-NM-125-AD, 1601 Lind Avenue SW., Renton, Washington 98055-4056.

The service information referenced in this AD may be obtained from Jetstream Aircraft, Inc., P.O. Box 16029, Dulles International Airport, Washington, DC 20041-6029. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Gregory Dunn, Aerospace Engineer,

Standardization Branch, ANM-113, FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington 98055-4056; telephone (206) 227-2799; fax (206) 227-1149.

SUPPLEMENTARY INFORMATION: The Civil Aviation Authority (CAA), which is the airworthiness authority for the United Kingdom, recently notified the FAA that an unsafe condition may exist on certain Jetstream Model 4101 airplanes. The CAA advises that reports have indicated that certain passenger seat belt shackles have failed. The reports indicate that the fork end of the shackles had become splayed, which contributed to the failures. This condition, if not corrected, could reduce the effectiveness of the passenger safety restraint system during take-off and landing or in an emergency situation, due to failure of the passenger seat belt shackles.

Jetstream has issued Alert Service Bulletin J41-A25-061, dated June 6, 1995, which describes procedures for a visual inspection of the seat belt assemblies to determine if certain shackles have been installed, and replacement of those seat belt shackles with improved ones. The improved shackles are not subject to the reported failure problems. The CAA classified this service bulletin as mandatory in order to assure the continued airworthiness of these airplanes in the United Kingdom.

This airplane model is manufactured in the United Kingdom and is type certificated for operation in the United States under the provisions of § 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the CAA has kept the FAA informed of the situation described above. The FAA has examined the findings of the CAA, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, this AD is being issued to prevent reduced effectiveness of the passenger safety restraint system during take-off and landing or in an emergency situation, due to failure of the passenger seat belt assemblies. This AD requires a visual inspection of the passenger seat belt assemblies to determine if certain seat belt shackles are installed, and replacement of those shackles with improved ones. The actions are required to be accomplished in accordance with