

Torque the strut ends and wing root bolts using adequate torque (do not over torque the attach fittings).

10. If evidence of intergranular corrosion is detected, remove and replace the corroded part with an airworthy part.

11. Upon completion of the inspection, replace the wing root fairings, wing inspection hole covers and wing strut covers.

Issued in Kansas City, Missouri on November 25, 1997.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 97-31680 Filed 12-2-97; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-CE-22-AD; Amendment 39-10225; AD 97-25-02]

RIN 2120-AA64

Airworthiness Directives; Mitsubishi Heavy Industries MU-2B Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to all Mitsubishi Heavy Industry (Mitsubishi) MU-2B series airplanes. This AD requires amending the Limitations Section of the airplane flight manual (AFM) to prohibit the positioning of the power levers below the flight idle stop while the airplane is in flight. This AFM amendment will include a statement of consequences if the limitation is not followed. This AD results from numerous incidents and five documented accidents involving airplanes equipped with turboprop engines where the propeller beta was improperly utilized during flight. The actions specified by this AD are intended to prevent loss of airplane control or engine overspeed with consequent loss of engine power caused by the power levers being positioned below the flight idle stop while the airplane is in flight.

EFFECTIVE DATE: January 21, 1998.

ADDRESSES: Information related to this AD may be examined at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97-CE-22-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

FOR FURTHER INFORMATION CONTACT: William Schinstock, Aerospace

Engineer, Wichita Aircraft Certification Office, FAA, 1801 Airport Road, Wichita, Kansas 67209; telephone (316) 946-4162; facsimile (316) 946-4407.

SUPPLEMENTARY INFORMATION:

Events Leading to the Issuance of This AD

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to all Mitsubishi Heavy Industry (Mitsubishi) MU-2B series airplanes was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on July 2, 1997 (62 FR 35696).

The NPRM proposed to require amending the Limitations Section of the AFM to prohibit the positioning of the power levers below the flight idle stop while the airplane is in flight, including a statement of consequences if the limitation is not followed. This AFM amendment shall consist of the following language:

Positioning of power levers below the flight idle stop while the airplane is in flight is prohibited. Such positioning may lead to loss of airplane control or may result in an overspeed condition and consequent loss of engine power.

The NPRM was the result of numerous incidents and five documented accidents involving airplanes equipped with turboprop engines where the propeller beta was improperly utilized during flight.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the two comments received from the manufacturer, Mitsubishi Heavy Industries, Inc.

Comment Issue No. 1: Change the "Explanation of the Provisions of the Proposed AD" Section of the NPRM

Mitsubishi explains that the statement "Since an unsafe condition has been identified that could exist or develop on other Mitsubishi MU-2B airplanes of the same type design," is misleading in that it leads the reader to believe that there is a design flaw with the MU-2B series airplanes. Mitsubishi includes proposed language to replace this phrase.

The FAA concurs that this statement could be misleading. This language is not repeated in the final rule so therefore no change is needed at this time. The FAA will keep Mitsubishi's comments in mind while drafting future AD's. No changes have been made to the final rule as a result of this comment.

Comment Issue No. 2: The Model MU-2B-26A Excluded From the NPRM

Mitsubishi states that the Model MU-2B-26A airplanes are excluded from the NPRM, and asks if this was an oversight on the FAA's part. Mitsubishi feels that these airplanes should be included in the AD.

Mitsubishi is correct in assuming that excluding the Model MU-2B-26A airplanes from the NPRM was an oversight. To add these airplanes in this rulemaking action would require the FAA to reopen the comment period and delay final rule action for all of the MU-2B series airplanes. The FAA will address the Model MU-2B-26A airplanes in a future rulemaking action. No changes have been made to the final rule as a result of this comment.

The FAA's Determination

After careful review of all available information related to the subject presented above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. The FAA has determined that these minor corrections will not change the meaning of the AD and will not add any additional burden upon the public than was already proposed.

Compliance Time of This AD

The FAA has determined that the compliance time of this AD should be specified in calendar time instead of hours time-in-service. While the condition addressed by this AD is unsafe while the airplane is in flight, the condition is not a result of repetitive airplane operation; the potential of the unsafe condition occurring is the same on the first flight as it is for subsequent flights. The compliance time of "30 days after the effective date of this AD" will not inadvertently ground airplanes and would assure that all owners/operators of the affected airplanes accomplish this AD in a reasonable time period.

Cost Impact

The FAA estimates that 437 airplanes in the U.S. registry will be affected by this AD, that it will take approximately 1 workhour per airplane to incorporate the required AFM amendment, and that the average labor rate is approximately \$60 an hour. Since an owner/operator who holds at least a private pilot's certificate can accomplish this AD, as authorized by sections 43.7 and 43.9 of the Federal Aviation Regulations (14 CFR 43.7 and 43.9), the only cost impact upon the public is the time it will take the affected airplane owner/operators to amend the AFM.

Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

97-25-02 Mitsubishi Heavy Industries:

Amendment 39-10225; Docket No. 97-CE-22-AD.

Applicability: Models MU-2B, MU-2B-10, MU-2B-15, MU-2B-20, MU-2B-25, MU-2B-26, MU-2B-30, MU-2B-35, MU-2B-36, MU-2B-36A, MU-2B-40, and MU-2B-60 airplanes, all serial numbers, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the

requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 30 days after the effective date of this AD, unless already accomplished.

To prevent loss of airplane control or engine overspeed with consequent loss of engine power caused by the power levers being positioned below the flight idle stop while the airplane is in flight, accomplish the following:

(a) Amend the Limitations Section of the airplane flight manual (AFM) by inserting the following language:

Positioning of power levers below the flight idle stop while the airplane is in flight is prohibited. Such positioning may lead to loss of airplane control or may result in an overspeed condition and consequent loss of engine power.

(b) This action may be accomplished by incorporating a copy of this AD into the Limitations Section of the AFM.

(c) Amending the AFM, as required by this AD, may be performed by the owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7), and must be entered into the aircraft records showing compliance with this AD in accordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(e) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Wichita Aircraft Certification Office (ACO), FAA, 1801 Airport Road, Wichita, Kansas. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita ACO.

(f) Information related to this AD may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

(g) This amendment (39-10225) becomes effective on January 21, 1998.

Issued in Kansas City, Missouri, on November 25, 1997.

Michael Gallagher,
Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 97-31675 Filed 12-2-97; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 97-CE-33-AD; Amendment 39-10224; AD 97-25-01]

RIN 2120-AA64

Airworthiness Directives; Raytheon Aircraft Company 58, 60, 90, 100, 200, and 300 Series and Model 2000 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to all Raytheon Aircraft Company (Raytheon) 58, 60, 90, 100, 200, and 300 series and Model 2000 airplanes (formerly referred to as Beech 58, 60, 90, 100, 200, and 300 series and Beech Model 2000 airplanes). This AD requires replacing certain AlliedSignal Aerospace outflow/safety valves in the pressurization system with new or serviceable valves. The AD results from a report of cracking and consequent failure of the affected outflow safety valves in the pressurization system. Investigation has revealed problems during the manufacturing process of certain AlliedSignal outflow/safety valves. The actions specified by this AD are intended to prevent outflow/safety valve cracking and consequent failure, which could result in rapid decompression of the airplane.

EFFECTIVE DATE: January 11, 1998.

ADDRESSES: Service information that applies to the proposed AD may be obtained from AlliedSignal Aerospace, Technical Publications, Department 65-70, P.O. Box 52170, Phoenix, Arizona 85072-2170. This information also may be examined at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97-CE-33-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

FOR FURTHER INFORMATION CONTACT: Michael D. Imbler, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; telephone (316) 946-4147; facsimile (316) 946-4407.

SUPPLEMENTARY INFORMATION:**Events Leading to the Issuance of This AD**

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would