

TABLE.—COMPLIANCE TIME FOR TLMCS—Continued

Condition of brackets and gaps	Compliance time
No crack in any flap vane bracket, but shims added	For Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A and CL-601-3R) series airplanes: Repeat inspections remain at 600 landings from rework. For Model CL-600-2B16 (CL-604) series airplanes: Repeat inspections remain at 1,800 landings from rework.
All 12 flap vane brackets have been replaced	For Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A and CL-601-3R) series airplanes: New threshold of 7,000 landings from installation of new flap vane brackets. Repeat inspections remain at 600 landings. For Model CL-600-2B16 (CL-604) series airplanes: New threshold of 7,200 landings from installation of new flap vane brackets. Repeat inspections remains at 1,800 landings.

(i) After doing the requirements of paragraph (h) of this AD, except as provided in paragraph (j) of this AD, no alternative inspection times may be approved for these flap vane brackets.

Alternative Methods of Compliance

(j) In accordance with 14 CFR 39.19, the Manager, New York ACO, FAA, is authorized to approve alternative methods of compliance (AMOCs) for this AD.

Incorporation by Reference

(k) Unless otherwise specified in this AD, the actions shall be done in accordance with the following Bombardier alert service bulletins as listed in Table 6 of this AD, as applicable. Table 6 of this AD is as follows:

TABLE 6.—ALERT SERVICE BULLETINS

Bombardier alert service bulletin	Excluding
A600-0699, Revision 01, dated July 8, 2002	Service Bulletin Incorporation Sheet, Flap Vane Bracket Inspection Program page, and Minimum Edge Distance Inspection pages.
A601-0532, Revision 01, dated July 8, 2002	Service Bulletin Incorporation Sheet, Flap Vane Bracket Inspection Program page, and Minimum Edge Distance Inspection pages
A604-27-007, Revision 01, dated July 8, 2002	Service Bulletin Incorporation Sheet, Flap Vane Bracket Inspection Program page, and Minimum Edge Distance Inspection pages.

This incorporation by reference was previously approved by the Director of the Federal Register as of May 8, 2003 (68 FR 19940, April 23, 2003). Copies of the service bulletins may be obtained from Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, suite 410, Westbury, New York; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Note 3: The subject of this AD is addressed in Canadian airworthiness directives CF-2002-36 and CF-2002-37, both effective August 30, 2002.

Effective Date

(1) This amendment becomes effective on June 11, 2004.

Issued in Renton, Washington, on April 28, 2004.

Kevin M. Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 04-10375 Filed 5-6-04; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NE-46-AD; Amendment 39-13557; AD 2004-07-13]

RIN 2120-AA64

Airworthiness Directives; General Electric Company CF6-80C2 Series Turbofan Engines; Correction

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document makes a correction to Airworthiness Directive (AD) 2004-07-13. That AD applies to General Electric Company (GE) CF6-80C2 series turbofan engines. That AD was published in the **Federal Register** on April 1, 2004 (69 FR 17033). The amendatory text in the Applicability section is incorrect. This document corrects the aircraft models that these engines are installed on. In all other respects, the original document remains the same.

DATES: *Effective Date:* Effective April 1, 2004.

FOR FURTHER INFORMATION CONTACT:

Karen Curtis, Aerospace Engineer, Aircraft Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; telephone (781) 238-7192; fax (781) 238-7199.

SUPPLEMENTARY INFORMATION: A final rule AD, FR Doc. 04-7235, that applies to General Electric Company (GE) CF6-80C2 series turbofan engines was published in the **Federal Register** on April 1, 2004 (69 FR 17033). The following correction is needed:

§ 39.13 [Corrected]

■ On page 17034, in the first column, in the Amendatory Section, Applicability paragraph (c), in the eighth line, "A300 and A330" is corrected to read "A300 and A310".

Issued in Burlington, MA, on May 3, 2004.

Jay J. Pardee,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 04-10429 Filed 5-6-04; 8:45 am]

BILLING CODE 4910-13-P