part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### **Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

## List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## **Adoption of the Amendment**

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

## § 39.13 [Amended]

■ 2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

## 2006–02–02 Empresa Brasileira De Aeronautica S.A. (EMBRAER):

Amendment 39–14454. Docket No. FAA–2005–22871; Directorate Identifier 2005–NM–191–AD.

#### Effective Date

(a) This AD becomes effective February 24, 2006.

## Affected ADs

(b) None.

# Applicability

(c) This AD applies to EMBRAER Model EMB-120, -120ER, -120FC, -120QC, and -120RT airplanes, certificated in any category; as identified in EMBRAER Service Bulletin 120–53–0080, dated November 30, 2004.

### **Unsafe Condition**

(d) This AD results from a report indicating that, during production, a pinhole was left open at the upper frame of the auxiliary power unit (APU) firewall. We are issuing this AD to ensure that the APU compartment is isolated from the rest of the airplane in the event of an APU fire.

#### Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

### **Rivet Installation**

(f) Within 4,000 flight hours or 24 months, whichever occurs first after the effective date of this AD: Install a rivet and washer in the hole of the upper frame of the APU firewall, in accordance with EMBRAER Service Bulletin 120–53–0080, dated November 30, 2004.

# Alternative Methods of Compliance (AMOCs)

(g)(1) The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office

### **Related Information**

(h) Brazilian airworthiness directive 2005–08–03, effective September 5, 2005, also addresses the subject of this AD.

# Material Incorporated by Reference

(i) You must use EMBRAER Service Bulletin 120-53-0080, dated November 30, 2004, to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of this document in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Empresa Brasileira de Aeronautica S.A. (EMBRAER), P.O. Box 343—CEP 12.225, Sao Jose dos Campos—SP, Brazil, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, Nassif Building, Washington, DC; on the Internet at http://dms.dot.gov; or at the National

Archives and Records Administration (NARA).

For information on the availability of this material at the NARA, call (202) 741–6030, or go to http://www.archives.gov/federal\_register/code\_of\_federal\_regulations/ibr\_locations.html.

Issued in Renton, Washington, on January 10, 2006.

### Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 06–469 Filed 1–19–06; 8:45 am] BILLING CODE 4910–13–P

## **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

### 14 CFR Part 39

[Docket No. FAA-2005-22873; Directorate Identifier 2005-NM-197-AD; Amendment 39-14457; AD 2006-02-05]

### RIN 2120-AA64

# Airworthiness Directives; Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Bombardier Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes. This AD requires replacing the Camloc fasteners on the sidewall of the center pedestal. This AD results from reports of the Camloc fasteners on the sidewall of the center pedestal disengaging and interfering with an inboard rudder pedal. We are issuing this AD to prevent these fasteners from disengaging and interfering with an inboard rudder pedal, which could reduce directional controllability of the airplane.

**DATES:** This AD becomes effective February 24, 2006.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of February 24, 2006.

ADDRESSES: You may examine the AD docket on the Internet at http://dms.dot.gov or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Nassif Building, room PL–401, Washington, DC.

Contact Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada, for service information identified in this AD.

### FOR FURTHER INFORMATION CONTACT:

Daniel Parrillo, Aerospace Engineer, Systems and Flight Test Branch, ANE– 172, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, suite 410, Westbury, New York 11590; telephone (516) 228–7305; fax (516) 794–5531.

### SUPPLEMENTARY INFORMATION:

## **Examining the Docket**

You may examine the airworthiness directive (AD) docket on the Internet at http://dms.dot.gov or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647–5227) is located on the plaza level of the Nassif Building at the street address stated in the ADDRESSES section.

#### Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain Bombardier Model CL—600—2B19 (Regional Jet Series 100 & 440) airplanes. That NPRM was published in the **Federal Register** on November 15, 2005 (70 FR 69291). That NPRM proposed to require replacing the Camloc fasteners on the sidewall of the center pedestal.

## Comments

We provided the public the opportunity to participate in the development of this AD. We received no comments on the NPRM or on the determination of the cost to the public.

## Conclusion

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

## **Costs of Compliance**

This AD will affect about 718 airplanes of U.S. registry. The required actions will take about 1 work hour per airplane, at an average labor rate of \$65 per work hour. Required parts will cost about \$135 per airplane. Based on these figures, the estimated cost of the AD for U.S. operators is \$143,600, or \$200 per airplane.

# **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

# **Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

# List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

## § 39.13 [Amended]

■ 2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

2006–02–05 Bombardier, Inc. (Formerly Canadair): Amendment 39–14457.

Docket No. FAA-2005-22873; Directorate Identifier 2005-NM-197-AD.

#### **Effective Date**

(a) This AD becomes effective February 24, 2006.

### Affected ADs

(b) None.

### **Applicability**

(c) This AD applies to Bombardier Model CL–600–2B19 (Regional Jet Series 100 & 440) airplanes, certificated in any category, serial numbers 7003 through 7986 inclusive.

#### Unsafe Condition

(d) This AD results from reports of the Camloc fasteners on the sidewall of the center pedestal disengaging and interfering with an inboard rudder pedal. We are issuing this AD to prevent these fasteners from disengaging and interfering with an inboard rudder pedal, which could reduce directional controllability of the airplane.

## Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

### **Replacement of Fasteners**

(f) Within 5,500 flight hours after the effective date of this AD, replace, with screws and nut plate assemblies, the Camloc fasteners on the left and right sidewalls of the center pedestal, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 601R–31–030, Revision F, dated September 1, 2005.

### **Actions Accomplished Previously**

(g) Replacing fasteners before the effective date of this AD in accordance with the Accomplishment Instructions of one of the issues of Bombardier Service Bulletin 601R–31–030 identified in Table 1 of this AD is acceptable for compliance with the requirements of paragraph (f) of this AD.

# TABLE 1.—PREVIOUS SERVICE BUL-LETIN REVISIONS ACCEPTABLE FOR COMPLIANCE

Issue of Bombardier Service Bulletin 601R–31–030	Date
Original	June 22, 2004. October 6, 2004. November 4, 2004. December 15, 2004. June 16, 2005. July 7, 2005.

# Alternative Methods of Compliance (AMOCs)

- (h)(1) The Manager, New York Aircraft Certification Office, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.
- (2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the

appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

### **Related Information**

(i) Canadian airworthiness directive CF–2004–23R1, dated July 18, 2005, also addresses the subject of this AD.

### Material Incorporated by Reference

(i) You must use Bombardier Service Bulletin 601R-31-030, Revision F, dated September 1, 2005, to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of this document in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, Nassif Building, Washington, DC; on the Internet at http:// dms.dot.gov; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to http://www.archives.gov/ federal\_register/code\_of\_federal\_regulations/ ibr\_locations.html.

Issued in Renton, Washington, on January 10, 2006.

### Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 06–466 Filed 1–19–06; 8:45 am] BILLING CODE 4910–13–P

# **DEPARTMENT OF TRANSPORTATION**

## **Federal Aviation Administration**

# 14 CFR Part 39

[Docket No. FAA-2005-22917; Directorate Identifier 2005-NM-157-AD; Amendment 39-14456; AD 2006-02-04]

### RIN 2120-AA64

Airworthiness Directives; Bombardier Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A, CL-601-3R, and CL-604) Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Bombardier Model CL–600–1A11 (CL–600), CL–600–2A12 (CL–601), and CL–600–2B16 (CL–601–3A, CL–601–3R, & CL–604) airplanes. This AD requires modifying the rudder balance spring assembly by installing a new adjustable

balance spring, and rigging the assembly to suit the rudder of each airplane. This AD results from production inspections that showed that the spring assembly that controls rudder balance may not have the correct pre-load on some airplanes. We are issuing this AD to prevent uncommanded yaw movements and consequent reduced controllability of the airplane.

**DATES:** This AD becomes effective February 24, 2006.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of February 24, 2006.

ADDRESSES: You may examine the AD docket on the Internet at http://dms.dot.gov or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, room PL-401, Washington, DC.

Contact Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada, for service information identified in this AD.

### FOR FURTHER INFORMATION CONTACT:

Daniel Parrillo, Aerospace Engineer, Systems and Flight Test Branch, ANE– 172, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228–7305; fax (516) 794–5531.

## SUPPLEMENTARY INFORMATION:

## **Examining the Docket**

You may examine the airworthiness directive (AD) docket on the Internet at http://dms.dot.gov or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647–5227) is located on the plaza level of the Nassif Building at the street address stated in the ADDRESSES section.

## Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain Bombardier Model CL–600–1A11 (CL–600), CL–600–2A12 (CL–601), and CL–600–2B16 (CL–601–3A, CL–601–3R, and CL–604) airplanes. That NPRM was published in the **Federal Register** on November 10, 2005 (70 FR 68377). That NPRM proposed to require modifying the rudder balance spring assembly by installing a new adjustable balance spring, and rigging the assembly to suit the rudder of each airplane.

### **Comments**

We provided the public the opportunity to participate in the development of this AD. We received no comments on the NPRM or on the determination of the cost to the public.

# **Explanation of Change to Related Information**

We have revised paragraph (k) of this AD to include reference to Canadian airworthiness directive CF-2005-21R1, dated November 23, 2005, which was issued to revise the applicability of the affected airplanes. This change was reflected in the applicability of the proposed AD.

### Conclusion

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD with the change described previously. We have determined that this change will neither increase the economic burden on any operator nor increase the scope of the AD.

## **Costs of Compliance**

This AD will affect about 501 airplanes of U.S. registry. The required actions will take about 12 work hours per airplane, at an average labor rate of \$65 per work hour. Required parts will cost about \$1,749 per airplane. Based on these figures, the estimated cost of the AD for U.S. operators is \$1,267,029, or \$2,529 per airplane.

## **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### **Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will