

board of directors, and noted in the minutes.

(2) *Customer identification program.* Each bank is subject to the requirements of 31 U.S.C. 5318(l) and the implementing regulation jointly promulgated by the FDIC and the Department of the Treasury at 31 CFR 1020.220.

**PART 334—FAIR CREDIT REPORTING**

■ 3. The authority citation for part 334 continues to read:

**Authority:** 12 U.S.C. 1818, 1819 (Tenth) and 1831 p–1; 15 U.S.C. 1681a, 1681b, 1681c, 1681m, 1681s, 1681s–3, 1681t, 1681w, 6801 and 6805, Pub. L. 108–159, 117 Stat. 1952.

■ 4. In § 334.82, revise paragraph (c)(2)(i)(A) to read as follows:

**§ 334.82 Duties of users regarding address discrepancies.**

\* \* \* \* \*

(c) \* \* \*

(2) \* \* \*

(i) \* \* \*

(A) Obtains and uses to verify the consumer’s identity in accordance with the requirements of the Customer Identification Program (CIP) rules implementing 31 U.S.C. 5318(l) (31 CFR 1020.220);

\* \* \* \* \*

■ 5. In Appendix J to Part 334, revise Section III, paragraph (a) to read as follows:

**Appendix J to Part 334—Interagency Guidelines on Identity Theft Detection, Prevention, and Mitigation**

\* \* \* \* \*

**III. Detecting Red Flags**

\* \* \* \* \*

(a) Obtaining identifying information about, and verifying the identity of, a person opening a covered account, for example, using the policies and procedures regarding identification and verification set forth in the Customer Identification Program rules implementing 31 U.S.C. 5318(l) (31 CFR 1020.220); and

\* \* \* \* \*

Dated at Washington, DC, this 15th day of March 2011.

By order of the Board of Directors.  
Federal Deposit Insurance Corporation.

**Robert E. Feldman,**  
*Executive Secretary.*

[FR Doc. 2011–6460 Filed 3–17–11; 8:45 am]

**BILLING CODE 6714–01–P**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 25**

[Docket No. NM436; Special Conditions No. 25–421–SC]

**Special Conditions: Boeing Model 747–8 Airplanes, Systems and Data Networks Security—Isolation or Protection From Unauthorized Passenger Domain Systems Access**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final special conditions.

**SUMMARY:** These special conditions are issued for Boeing Model 747–8 airplanes. This airplane will have novel or unusual design features associated with connectivity of the passenger domain computer systems to the airplane critical systems and data networks. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for these design features. These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

**DATES:** *Effective Date:* April 18, 2011  
**FOR FURTHER INFORMATION CONTACT:** Will Struck, FAA, Airplane and Flight Crew Interface Branch, ANM–111, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–2764; facsimile (425) 227–1149.

**SUPPLEMENTARY INFORMATION:**

**Background**

On November 4, 2005, The Boeing Company, P.O. Box 3707, Seattle, WA 98124, applied for an amendment to Type Certificate Number A20WE to include the new Model 747–8 passenger airplane. The Model 747–8 is a derivative of the 747–400. The Model 747–8 is a four-engine jet transport airplane that will have a maximum takeoff weight of 970,000 pounds and new General Electric GENx-2B67 engines. The Model 747–8 will have two flight crew and the capacity to carry 605 passengers.

**Type Certification Basis**

Under the provisions of Title 14 Code of Federal Regulations (14 CFR) 21.101, Boeing must show that the Model 747–8 (hereafter referred to as the 747–8) continues to meet the applicable provisions of 14 CFR part 25, as amended by Amendments 25–1 through 25–117, except for §§ 25.809 and 25.812,

which will remain at Amendment 25–115. These regulations will be incorporated into Type Certificate No. A20WE after type certification approval of the 747–8.

If the Administrator finds that the applicable airworthiness regulations (i.e., part 25) do not contain adequate or appropriate safety standards for the 747–8 because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

Special conditions are initially applicable to the model for which they are issued. Should the type certificate for that model be amended later to include any other model that incorporates the same novel or unusual design feature, or should any other model already included on the same type certificate be modified to incorporate the same novel or unusual design feature, the special conditions would also apply to the other model under § 21.101.

In addition to the applicable airworthiness regulations and special conditions, the 747–8 must comply with the fuel vent and exhaust emission requirements of 14 CFR part 34 and the noise certification requirements of 14 CFR part 36.

Special conditions, as defined in § 11.19, are issued under § 11.38, and become part of the type certification basis under § 21.101.

**Novel or Unusual Design Features**

The Boeing Model 747–8 airplane will incorporate the following novel or unusual design features: Digital systems architecture composed of several connected networks. The network architecture would be used for a diverse set of functions, including:

- 1. Flight-safety related control, communication, and navigation systems (Aircraft Control Domain),
- 2. Airline business and administrative support (Airline Information Domain),
- 3. Passenger information and entertainment systems (Passenger Entertainment Domain), and
- 4. The capability to allow access to or by external network sources.

**Discussion of Comments**

Notice of proposed special conditions No. 25–10–01–SC for the Boeing Model 747–8 airplane was published in the **Federal Register** on December 9, 2010 (75 FR 76647). No comments were received and the special conditions are adopted as proposed.

**Applicability**

As discussed above, these special conditions are applicable to Boeing

Model 747-8 airplanes. Should Boeing apply at a later date for a change to the type certificate to include another model incorporating the same novel or unusual design features, the special conditions would apply to that model as well.

#### Conclusion

This action affects only certain novel or unusual design features of the Boeing Model 747-8 airplane. It is not a rule of general applicability.

#### List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

#### The Special Conditions

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for the Boeing Model 747-8 airplane.

The design must prevent all inadvertent or malicious changes to, and all adverse impacts upon, all systems, networks, hardware, software, and data in the Aircraft Control Domain and in the Airline Information Domain from all points within the Passenger Information and Entertainment Domain.

Issued in Renton, Washington, on March 9, 2011.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate,  
Aircraft Certification Service.*

[FR Doc. 2011-6323 Filed 3-17-11; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 25

[Docket No. NM437; Special Conditions No. 25-422-SC]

#### Special Conditions: Gulfstream Model GVI Airplane; Electronic Flight Control System Mode Annunciation.

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final special conditions.

**SUMMARY:** These special conditions are issued for the Gulfstream GVI airplane. This airplane will have novel or unusual design features when compared to the state of technology envisioned in the airworthiness standards for transport category airplanes. These design features include an electronic

flight control system. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for these design features. These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

**DATES:** *Effective Date:* April 18, 2011.

**FOR FURTHER INFORMATION CONTACT:** Joe Jacobsen, FAA, Airplane and Flight Crew Interface Branch, ANM-111, Transport Standards Staff, Transport Airplane Directorate, Aircraft Certification Service, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2011; facsimile (425) 227-1320.

#### SUPPLEMENTARY INFORMATION:

##### Background

On March 29, 2005, Gulfstream Aerospace Corporation (hereafter referred to as "Gulfstream") applied for an FAA type certificate for its new Gulfstream Model GVI passenger airplane. Gulfstream later applied for, and was granted, an extension of time for the type certificate, which changed the effective application date to September 28, 2006. The Gulfstream Model GVI airplane will be an all-new, two-engine jet transport airplane with an executive cabin interior. The maximum takeoff weight will be 99,600 pounds, with a maximum passenger count of 19 passengers.

##### Type Certification Basis

Under provisions of Title 14 Code of Federal Regulations (14 CFR) 21.17, Gulfstream must show that the Gulfstream Model GVI airplane (hereafter referred to as "the GVI") meets the applicable provisions of 14 CFR part 25, as amended by Amendments 25-1 through 25-119, 25-122 and 25-124. If the Administrator finds that the applicable airworthiness regulations (*i.e.*, 14 CFR part 25) do not contain adequate or appropriate safety standards for the GVI because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

Special conditions are initially applicable to the model for which they are issued. Should the type certificate for that model be amended later to include any other model that incorporates the same novel or unusual design features, the special conditions would also apply to the other model under the provisions of § 21.101.

In addition to complying with the applicable airworthiness regulations and special conditions, the GVI must

comply with the fuel vent and exhaust emission requirements of 14 CFR part 34 and the noise certification requirements of 14 CFR part 36. The FAA must also issue a finding of regulatory adequacy pursuant to section 611 of Public Law 92-574, the "Noise Control Act of 1972."

The FAA issues special conditions, as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type certification basis under § 21.17(a)(2).

##### Novel or Unusual Design Features

The GVI will have a fly-by-wire electronic flight control system. This system provides an electronic interface between the pilot's flight controls and the flight control surfaces for both normal and failure states, and it generates the actual surface commands that provide for stability augmentation and control about all three airplane axes. Because electronic flight control system technology has outpaced existing regulations (primarily §§ 25.671 and 25.672), a special condition is needed to ensure appropriate mode recognition by the flight crew for events which significantly change the operating mode of the electronic flight control system.

##### Discussion of Comments

Notice of proposed special conditions No. 25-10-02-SC for Gulfstream GVI airplanes was published in the **Federal Register** on December 13, 2010 (75 FR 77569). Only one comment was received.

##### Clarification of Conditions That Should Be Announced

The commenter, Gulfstream, requested that the special conditions be revised to clarify the conditions in which the mode annunciation should occur. Gulfstream suggested that additional annunciation should not be required when transitioning from one normal operation mode to another in response to flight crew actions, such as extending flaps or landing gear.

We do not agree with the commenter's recommendation. The current verbiage clearly states that the mode annunciation is only required when "normal handling or operational characteristics" of the airplane are significantly changed or degraded. In the scenario that Gulfstream refers to, there would be no change to the "normal handling or operational characteristics." Therefore, no annunciation would be required. No changes were made as a result of this comment and the special conditions are adopted as proposed.