

(b) The credit union must deliver the notice in person to each member, or mail it to each member at the address for the member as it appears on the records of the credit union, at least 45 days, but not more than 90 days, before the date of the vote. Members must be permitted to vote by mail ballot. The credit union may provide the notice of the proposal and the ballot to members at the same time.

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■ 11. Revise § 708b.204(b) to read as follows:

§ 708b.204 Notice to members of proposal to convert insurance.

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(b) The credit union must deliver the notice in person to each member, or mail it to each member at the address for the member as it appears on the records of the credit union, at least 45 days, but not more than 90 days, before the date of the vote. Members must be permitted to vote by mail ballot. The credit union may provide the notice of the proposal and the ballot to members at the same time.

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[FR Doc. 2017-11331 Filed 6-7-17; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2017-0254; Directorate Identifier 2017-NE-10-AD]

RIN 2120-AA64

Airworthiness Directives; General Electric Company Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all General Electric Company (GE) CF34-8E model turbofan engines. This proposed AD was prompted by a report that using a certain repair procedure for the fan outlet guide vane (OGV) frame could alter the strength capability of the fan OGV frame. This proposed AD would require replacement of all fan OGV frames repaired using this procedure.

We are proposing this AD to correct the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by July 24, 2017.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** 202-493-2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- **Hand Delivery:** Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact General Electric Company, GE-Aviation, Room 285, 1 Neumann Way, Cincinnati, OH 45215, phone: 513-552-3272; fax: 513-552-3329; email: geae.aoc@ge.com. You may view this service information at the FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA. For information on the availability of this material at the FAA, call 781-238-7125.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0254; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (phone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Martin Adler, Aerospace Engineer, Engine Certification Office, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781-238-7157; fax: 781-238-7199; email: martin.adler@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about

this proposal. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2017-0254; Directorate Identifier 2017-NE-10-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

We received a report that using a certain repair procedure for the fan OGV frame could alter the strength capability of the fan OGV frame because the repair procedure included an improper heat cycle. This proposed AD would require replacement of all fan OGV frames repaired using this procedure. This condition, if not corrected, could result in failure of the fan OGV frame, engine separation, and loss of the airplane.

Related Service Information

We reviewed GE CF34-8E Engine Manual, GEK 112031, 72-00-23, REPAIR 006. The repair describes procedures for applying a dry-film lubricant to the fan OGV frame with heat curing.

FAA’s Determination

We are proposing this AD because we evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

Proposed AD Requirements

This proposed AD would require replacement of fan OGV frames.

Costs of Compliance

We estimate that this proposed AD affects 42 engines installed on airplanes of U.S. registry.

We estimate the following costs to comply with this proposed AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Fan OGV frame part—annual, prorated cost	0 work-hour × \$85 per hour = \$0.00	\$12,300.00	\$12,300.00	\$516,600.00

According to the manufacturer, some of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected individuals. We do not control warranty coverage for affected individuals. As a result, we have included all costs in our cost estimate.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
- (3) Will not affect intrastate aviation in Alaska, and
- (4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

General Electric Company: Docket No. FAA–2017–0254; Directorate Identifier 2017–NE–10–AD.

(a) Comments Due Date

We must receive comments by July 24, 2017.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all General Electric Company (GE) CF34–8E2; CF34–8E2A1; CF34–8E5; CF34–8E5A1; CF34–8E5A2; CF34–8E6; and CF34–8E6A1 model turbofan engines.

(d) Subject

Joint Aircraft System Component (JASC), 7270, Turbine Engine Bypass Section.

(e) Unsafe Condition

This AD was prompted by a report that using a certain repair procedure for the fan outlet guide vane (OGV) frame could alter the strength capability of the fan OGV frame. We are issuing this AD to prevent failure of the fan OGV frame, engine separation, and loss of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

- (1) For engines with a fan OGV frame installed that was repaired using GE CF34–8E Engine Manual, GEK 112031, 72–00–23, REPAIR 006:
 - (i) If the fan OGV frame has 24,900 cycles since new (CSN) or more on the effective date

of this AD, remove the OGV frame from service within 100 cycles after the effective date of this AD.

(ii) If the OGV frame has less than 24,900 CSN on the effective date of this AD, remove the fan OGV frame from service at the next shop visit after the effective date of this AD, or before exceeding 25,000 CSN, whichever occurs earlier.

(2) After the effective date of this AD, do not install a fan OGV frame that was repaired using GE CF34–8E Engine Manual, GEK 112031, 72–00–23, REPAIR 006.

(h) Definition

For the purpose of this AD, an “engine shop visit” is the induction of an engine into the shop for maintenance involving the separation of pairs of major mating engine flanges.

(i) Alternative Methods of Compliance (AMOCs)

The Manager, Engine Certification Office, FAA, may approve AMOCs for this AD. Use the procedures found in 14 CFR 39.19 to make your request. You may email your request to: ANE-AD-AMOC@faa.gov.

(j) Related Information

(1) For more information about this AD, contact Martin Adler, Aerospace Engineer, Engine Certification Office, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781–238–7157; fax: 781–238–7199; email: martin.adler@faa.gov.

(2) GE CF34–8E Engine Manual, GEK 112031, 72–00–23, REPAIR 006 can be obtained from GE using the contact information in paragraph (j)(3) of this AD.

(3) For service information identified in this AD, contact General Electric Company, GE-Aviation, Room 285, 1 Neumann Way, Cincinnati, OH 45215, phone: 513–552–3272; fax: 513–552–3329; email: geae.aoc@ge.com.

(4) You may view this service information at the FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA. For information on the availability of this material at the FAA, call 781–238–7125.

Issued in Burlington, Massachusetts, on May 25, 2017.

Carlos A. Pestana,

Acting Assistant Manager, Engine & Propeller Directorate, Aircraft Certification Service.

[FR Doc. 2017–11782 Filed 6–7–17; 8:45 am]

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