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DEPARTMENT OF AGRICULTURE

Office of the Secretary

7 CFR Part 1

[Docket No. USDA–2021–0009]

RIN 0503–AA74

Production or Disclosure of Official Information in Legal Proceedings

AGENCY: Office of the Secretary, USDA.

ACTION: Direct final rule; confirmation of effective date.

SUMMARY: On February 28, 2022, the U.S. Department of Agriculture published a direct final rule. The direct final rule notified the public of our intention to revise our regulations regarding the production or disclosure of official information in legal proceedings (referred to as *Touhy* regulations). We did not receive any substantive written adverse comments in response to the direct final rule.

DATES: The effective date of the direct final rule published February 28, 2022, at 87 FR 10925 is confirmed as April 29, 2022.

FOR FURTHER INFORMATION CONTACT: Karen Carrington Fletcher, Senior Counsel, Office of the General Counsel, USDA, 1400 Independence Ave. SW, Room 103–W, Washington, DC 20250; karen.fletcher@usda.gov; (202) 720–0944.

SUPPLEMENTARY INFORMATION: On February 28, 2022, the U.S. Department of Agriculture published a direct final rule. (87 FR 10925) The direct final rule notified the public of our intention to revise our regulations regarding the production or disclosure of official information in legal proceedings (referred to as *Touhy* regulations).

In the direct final rule, we stated that if we received no written adverse comments or written notice of intent to submit adverse comments within 30 days of publication of the direct final rule, the direct final rule would become

effective 60 days following its publication.

During the comment period, we received three comments, two of which were submitted by the same commenter. These three comments suggested that the amended regulations had the potential to obstruct justice and therefore should be withdrawn. However, the comments provided no basis for that assertion or any details that would allow us to understand the commenters' rationale. Therefore, we have determined that the regulations should become effective as stated in the direct final rule.

Accordingly, we are confirming that the direct final rule published February 28, 2022, will become effective on April 29, 2022.

Authority: 5 U.S.C. 301, unless otherwise noted.

David Grahn,

Principal Deputy General Counsel, United States Department of Agriculture.

[FR Doc. 2022–08814 Filed 4–25–22; 8:45 am]

BILLING CODE 3410–90–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2022–0022; Project Identifier AD–2020–01264–A; Amendment 39–22033; AD 2022–09–13]

RIN 2120–AA64

Airworthiness Directives; Piper Aircraft, Inc. Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Piper Aircraft, Inc. (Piper) Model PA–34–200 airplanes. This AD was prompted by the determination that the life limit for alternate bolts that attach the drag link to the nose gear were not listed as airworthiness limitations. This AD requires establishing a life limit for these bolts. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective May 31, 2022.

ADDRESSES: For service information identified in this final rule, contact Piper Aircraft, Inc., 2926 Piper Drive, Vero Beach, FL 32960; phone: (772) 299–2141; website: <https://www.piper.com>. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222–5110.

Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2022–0022; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Fred Caplan, Aviation Safety Engineer, Atlanta ACO Branch, FAA, 1701 Columbia Avenue, College Park, GA 30337; phone: (404) 474–5507; email: frederick.n.caplan@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain serial-numbered Piper Model PA–34–200 airplanes. The NPRM published in the **Federal Register** on February 3, 2022 (87 FR 6089; corrected February 16, 2022, 87 FR 8752). The NPRM was prompted by a notification from Piper that prior revisions of the airworthiness limitations section (ALS) for certain Piper Model PA–34–200 airplanes did not contain a life limit for bolt part number (P/N) 693–215 (standard P/N NAS6207–50D). Bolt P/N 693–215 (NAS6207–50D) is an alternate part for P/N 400–274 (standard P/N AN7–35). These bolts attach the drag link to the nose gear trunnion on Piper Model PA–34–200 airplanes. Piper did not include an ALS revision for the P/N 693–215 (standard P/N NAS6207–50D) bolt to establish the same life limit as the P/N 400–274 (AN7–35). If bolt P/N 693–215 (standard P/N NAS6207–

50D) that attaches the drag link to the nose gear trunnion remains in service beyond its fatigue life, failure of the nose landing gear could occur, which could result in loss of airplane control during take-off, landing, or taxi operations.

In the NPRM, the FAA proposed to require establishing a 500-hour life limit for bolt P/N 693–215 and P/N NAS6207–50D. The FAA is issuing this AD to address the unsafe condition on these products.

Discussion of Final Airworthiness Directive

Comments

The FAA received comments from two individual commenters. The following presents the comments received on the NPRM and the FAA’s response.

One individual supported the NPRM without change.

Another individual requested the FAA revise the proposed AD by requiring different assembly procedures and hardware as terminating action. The commenter stated that failure of the bolt results from the bolts not being tightened properly or loosening up in service. The commenter noted that this can be corrected with improved maintenance instructions to achieve the proper torque and hardware (thinner

washers, a longer bushing, or a slightly longer bolt, for example) to provide sufficient lateral clearance on the bushing to avoid binding.

The FAA disagrees with the commenter’s suggestion. This AD is not addressing potential failure of the bolt through maintenance practices but instead addresses the life limit for the subject bolt, which is part of the aircraft’s type design. The life limit was inadvertently omitted from the ALS, and this AD simply corrects that omission. To the extent the commenter requested a terminating action, this request is unnecessary as this AD only requires a one-time change to the aircraft maintenance records.

The FAA did not change this AD based on this comment.

Conclusion

The FAA reviewed the relevant data, considered the comments received, and determined that air safety requires adopting the AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on these products. This AD is adopted as proposed in the NPRM.

Related Service Information

The FAA reviewed Piper Seneca Service Manual, Airworthiness Limitations, 753–817, page 1–1, dated November 30, 2019. This service

information specifies the life limits of the P/N 693–215 (standard P/N NAS6207–50D) bolt that attaches the drag link to the nose gear trunnion.

ADs Mandating Airworthiness Limitations

The FAA has previously mandated airworthiness limitations by issuing ADs that require revising the ALS of the existing maintenance manual or instructions for continued airworthiness to incorporate new or revised inspections and life limits. This AD, however, requires incorporating new or revised inspections and life limits into the maintenance records required by 14 CFR 91.417(a)(2) or 135.439(a)(2) for your airplane. The FAA does not intend this as a substantive change. Requiring incorporation of the new ALS requirements into the maintenance records, rather than requiring individual repetitive inspections and replacements, allows operators to record AD compliance once after updating the maintenance records, rather than recording compliance after every inspection and part replacement.

Costs of Compliance

The FAA estimates that this AD affects 187 airplanes of U.S. registry.

The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per airplane	Cost on U.S. operators
Revise the Airworthiness Limitations	1 work-hour × \$85 per hour = \$85	Not Applicable	\$85	\$15,895

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:
Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2022–09–13 Piper Aircraft, Inc.:
 Amendment 39–22033; Docket No.

FAA-2022-0022; Project Identifier AD-2020-01264-A.

(a) Effective Date

This airworthiness directive (AD) is effective May 31, 2022.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Piper Aircraft, Inc. Model PA-34-200 airplanes, serial numbers 34-7250001 through 34-7450220, certificated in any category.

(d) Subject

Joint Aircraft System Component (JASC) Code 3220, Nose/Tail Landing Gear.

(e) Unsafe Condition

This AD was prompted by the determination that the life limit for alternate bolts that attach the drag link to the nose gear were not included as airworthiness limitations. The FAA is issuing this AD to establish a life limit on bolt part numbers 693-215 and NAS6207-50D that attach the drag link to the nose gear trunnion. The unsafe condition, if not addressed, could result in failure of the nose landing gear and lead to loss of airplane control during take-off, landing, or taxi operations.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Actions

(1) Within 90 days after the effective date of this AD, incorporate into the maintenance records required by 14 CFR 91.417(a)(2) or 135.439(a)(2) for your airplane a life limit of 500 hours for bolt part numbers 693-215 and NAS6207-50D.

Note to paragraph (g)(1): Piper Seneca Service Manual, Airworthiness Limitations, 753-817, page 1-1, dated November 30, 2019, contains the life limit in paragraph (g)(1) of this AD.

(2) Thereafter, except as provided in paragraph (h)(1) of this AD, no alternative replacement times may be approved for these bolts.

(h) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Atlanta ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (i)(1) of this AD.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(i) Related Information

(1) For more information about this AD, contact Fred Caplan, Aviation Safety

Engineer, Atlanta ACO Branch, FAA, 1701 Columbia Avenue, College Park, GA 30337; phone: (404) 474-5507; email: frederick.n.caplan@faa.gov.

(2) For service information identified in this AD that is not incorporated by reference, contact Piper Aircraft, Inc., 2926 Piper Drive, Vero Beach, FL 32960; phone: (772) 299-2141; website: <https://www.piper.com>. You may view this referenced service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (817) 222-5110.

(j) Material Incorporated by Reference

None.

Issued on April 21, 2022.

Gaetano A. Sciortino,

Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2022-08852 Filed 4-25-22; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2022-0145; Project Identifier MCAI-2021-00522-R; Amendment 39-22027; AD 2022-09-07]

RIN 2120-AA64

Airworthiness Directives; Bell Textron Canada Limited (Type Certificate Previously Held by Bell Helicopter Textron Canada Limited) Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2019-11-05 for certain Bell Helicopter Textron Canada Limited (now Bell Textron Canada Limited) Model 429 helicopters. AD 2019-11-05 required inspecting the tail rotor (TR) pitch link assemblies, and replacing certain pitch link bearings. This AD was prompted by a report of a worn pitch link, and the FAA's determination that all TR pitch link assemblies are affected by the unsafe condition. This AD continues to require the actions specified in AD 2019-11-05, and revises the applicability and requires inspections of certain other TR pitch link assemblies. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective May 31, 2022.

The Director of the Federal Register approved the incorporation by reference

of a certain publication listed in this AD as of May 31, 2022.

ADDRESSES: For service information identified in this final rule, contact Bell Textron Canada Limited, 12,800 Rue de l'Avenir, Mirabel, Quebec J7J 1R4, Canada; phone: 1-450-437-2862 or 1-800-363-8023; fax: 1-450-433-0272; email: productsupport@bellflight.com; or at <https://www.bellflight.com/support/contact-support>. You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222-5110. It is also available at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2022-0145.

Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2022-0145; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the Transport Canada Civil Aviation (TCCA) AD, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Hal Jensen, Aerospace Engineer, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 950 L'Enfant Plaza N SW, Washington, DC 20024; phone: (202) 267-9167; email: hal.jensen@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2019-11-05, Amendment 39-19651 (84 FR 26546, June 7, 2019) (AD 2019-11-05). AD 2019-11-05 applied to certain Bell Helicopter Textron Canada Limited Model 429 helicopters. The NPRM published in the **Federal Register** on February 23, 2022 (87 FR 10107). In the NPRM, the FAA proposed to continue to require inspecting the TR pitch link assemblies, and replacing certain pitch link bearings, as well as proposed to revise the applicability and require inspections of certain other TR pitch link assemblies. The NPRM was prompted by TCCA AD CF-2015-16R3, dated April 30, 2021 (TCCA AD CF-