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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 23

[Docket No. FAA–2024–0969]

Accepted Means of Compliance (MOC); Airworthiness Standards: Normal Category Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of availability (NOA).

SUMMARY: This document announces the availability of ASTM International (ASTM) consensus standards for use as an FAA-accepted means of compliance (MOC) to the applicable airworthiness standards for normal category airplanes. The FAA accepts ASTM F3264–23, “Standard Specification for Normal Category Aeroplanes Certification,” with changes identified in this document.

DATES: Acceptance of the means of compliance is effective on May 29, 2024.

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SUPPLEMENTARY INFORMATION:

Background

Under the provisions of the National Technology Transfer and Advancement Act of 1995¹ and Office of Management and Budget (OMB) Circular A–119, “Federal Participation in the Development and Use of Voluntary Consensus Standards and in Conformity Assessment Activities,” effective January 27, 2016, the FAA participates in the development of consensus standards for use as a means of carrying out its policy objectives where appropriate.

Consistent with the Small Airplane Revitalization Act of 2013, the FAA has been working with industry and other stakeholders through the ASTM F44 Committee on General Aviation Aircraft to develop consensus standards as a MOC in certifying small airplanes under part 23.²

In part 23, amendment 23–64 (81 FR 96572), published on December 30, 2016,³ the final rule stated the FAA would publish an NOA of those consensus standards in the **Federal Register** when the Administrator accepts the consensus standards as an acceptable MOC. The FAA reviewed the published ASTM consensus standards developed by the ASTM F44 Committee

¹ Ref Public Law 104–113 as amended by Public Law 107–107.

² Ref Public Law 113–53.

³ See <https://www.federalregister.gov/documents/2016/12/30/2016-30246/revision-of-airworthiness-standards-for-normal-utility-acrobatic-and-commuter-category-airplanes>.

on General Aviation Aircraft as the basis for a MOC to 65 sections of part 23, amendments 23–64 and 23–65. In some cases, the Administrator found sections of ASTM F3264–23, without changes, are accepted as an MOC for the airworthiness requirements of part 23, amendments 23–64 and 23–65. In other cases, the MOC, while based on ASTM consensus standards, include additional FAA provisions necessary to comply with the airworthiness requirements of part 23, amendments 23–64 and 23–65.

Part 23, amendment 23–64, established airworthiness requirements based on the safety requirements outlined in amendment 23–63, except in areas that address loss of control and icing, where the FAA increased the safety level. Depending on the details of a design, the applicant may require use of a different MOC beyond those accepted by this NOA. For example, novel airplane designs, such as unmanned airplanes or vertical takeoff and landing airplanes, may be outside the scope of this NOA, and applicants may need to propose alternative MOC applicable to their designs accepted under § 23.2010.

Means of Compliance Accepted

The FAA accepts only the revisions of the standards referenced in ASTM F3264–23, “Standard Specification for Normal Category Aeroplanes Certification,” as a means of compliance for part 23, amendments 23–64 and 23–65, with the changes identified in table 1. For ease of use, table 2 was added to provide a side-by-side view, linking the applicable part 23 regulation to the ASTM F3264–23 section.

TABLE 1—PART 23 ACCEPTED MOC BASED ON ASTM CONSENSUS STANDARDS ^{4 5}

ASTM No. as identified in F3264–23	ASTM document title	Changes required for FAA acceptance ⁴	Additional information ⁵
F2490–20	Standard Guide for Aircraft Electrical Load and Power Source Capacity Analysis.	None.	
F3061/F3061M–22b	Standard Specification for Systems and Equipment in Aircraft.	<p>The tables defining applicability of requirements in sections 4, 10, 13, and 17 are not accepted. Applicability will be determined by the Policy and Standards Division.</p> <p>Replace 17.3.1.1 with the following:</p> <p>(a) Each electrical or electronic system that performs a function, the failure of which would prevent the continued safe flight and landing of the airplane, must be designed, and installed such that—</p> <p>(1) The function at the airplane level is not adversely affected during and after the time the airplane is exposed to lightning; and</p> <p>(2) The system recovers normal operation of that function in a timely manner after the airplane is exposed to lightning unless the system's recovery conflicts with other operational or functional requirements of the system.</p> <p>Replace 17.3.1.2 with the following:</p> <p>(b) Each electrical and electronic system that performs a function, the failure of which would significantly reduce the capability of the airplane or the ability of the flight crew to respond to an adverse operating condition, must be designed and installed such that the system recovers normal operation of that function in a timely manner after the airplane is exposed to lightning.</p> <p>Remove 17.3.1.3.</p> <p>Replace 10.4.2, 10.4.3, 10.4.4 Level 1 with Level 4.</p>	<p>The tables defining applicability found in F3061/F3061M–22b sections 4, 10, 13, and 17 are not accepted. Applicability will be determined by the Policy and Standards Division. Aircraft certification levels are as defined in 14 CFR 23.2005.</p> <p>F3061/F3061M–22b does not contain means for showing compliance to §23.2310, <i>Buoyancy for seaplanes and amphibians</i>. If applying for certification of a seaplane or amphibian, applicants may use the provisions of §§23.751, 23.755, and 23.757 at amendment 23–63 as a means of complying with §23.2310 or may obtain FAA acceptance of a different MOC in accordance with §23.2010.</p>
F3062/F3062M–20	Standard Specification for Aircraft Powerplant Installation.	None.	
F3063/F3063M–21	Standard Specification for Aircraft Fuel Storage and Delivery.	None.	
F3064/F3064M–21	Standard Specification for Aircraft Powerplant Control, Operation, and Indication.	None.	
F3065/F3065M–21a	Standard Specification for Aircraft Propeller System Installation.	None.	
F3066/F3066M–18	Standard Specification for Aircraft Powerplant Installation Hazard Mitigation.	None.	
F3082/F3082M–22	Standard Specification for Weights and Centers of Gravity of Aircraft.	None.	
F3083/F3083M–20a	Standard Specification for Emergency Conditions, Occupant Safety and Accommodations.	<p>Replace 4.1.6 with the following:</p> <p>Powerplant and ESS mounts and supporting structures must withstand 18.0 g forward for powerplants and ESS installed behind and above the seating compartment.</p>	
F3093/F3093M–21	Standard Specification for Aeroelasticity Requirements.	None.	
F3114–21	Standard Specification for Structures.	None.	
F3115/F3115M–23	Standard Specification for Structural Durability for Small Aeroplanes.	None	<p>If the applicant proposes to use F3115/F3115M–20 sections 4.3 or 6.3.2, the Policy and Standards Division will be involved as the standard is applied during projects to review the approach to determining similarity (F3115/F3115M–20 section 4.3) and criteria defining obvious damage (F3115/F3115M–20 section 6.3.2).</p> <p>If the applicant proposes to use F3115/F3115M–23 sections 4.1.3 or 4.3.3.2, the Policy and Standards Division will be involved as the standard is applied during projects to review the approach to determining similarity (F3115/F3115M–23 section 4.1.3) and criteria defining obvious damage (F3115/F3115M–23 section 4.3.3.2).</p>

TABLE 1—PART 23 ACCEPTED MOC BASED ON ASTM CONSENSUS STANDARDS^{4 5}—Continued

ASTM No. as identified in F3264–23	ASTM document title	Changes required for FAA acceptance ⁴	Additional information ⁵
F3116/F3116M–18e2	Standard Specification for Design Loads and Conditions.	Replace: Section 4.1.4 with: “Appendix X1 through Appendix X4 provides, within the limitations specified within the appendix, a simplified MOC with several of the requirements set forth in sections 4.2 to 4.26 and 7.1 to 7.9 that can be applied as one (but not the only) means to comply. If the simplified methods in appendix X1 through X3 are used, they must be used together in their entirety.” Replace: Section X1.1.1 with: “The methods provided in this appendix provide one possible means (but not the only possible means) of compliance and can only be applied to Level 1 and Level 2 low speed airplanes.” Replace: Section X2.1.1 with: “The methods provided in this appendix provide one possible means (but not the only possible means) of compliance and can only be applied to Level 1 and Level 2 low speed airplanes.” Replace: Section X3.1.1 with: “The methods provided in this appendix provide one possible means (but not the only possible means) of compliance and can only be applied to Level 1 and Level 2 low speed airplanes.” Replace: Section X4.1.1 with: “The methods provided in this appendix provide one possible means (but not the only possible means) of compliance and can only be applied to Level 1 low speed airplanes.”	
F3117/F3117M–23a	Standard Specification for Crew Interface in Aircraft.	None.	
F3120/F3120M–20	Standard Specification for Ice Protection for General Aviation Aircraft.	None.	
F3173/F3173M–21a	Standard Specification for Aircraft Handling Characteristics.	None.	
F3174/F3174M–21	Standard Specification for Establishing Operating Limitations and Information for Aeroplanes.	None.	
F3179/F3179M–22e1	Standard Specification for Performance of Aircraft.	None.	
F3180/F3180M–21	Standard Specification for Low-Speed Flight Characteristics of Aircraft.	The FAA does not universally accept all alternatives contained in F3180/F3180M–21. FAA previously and continues to accept F3180/F3180M–16.	Applicants are encouraged to propose elements of F3180/F3180M–21, or other means, for development of their project MOC to § 23.2150 in accordance with § 23.2010.
F3227/F3227M–22	Standard Specification for Environmental Systems in Aircraft.	None.	
F3228–21	Standard Specification for Flight Data and Voice Recording in Small Aircraft.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3228–21 are not accepted. Applicability will be determined by the Policy and Standards Division.
F3229/F3229M–17	Standard Practice for Static Pressure System Tests in Small Aircraft.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3229/F3229M–17 is not accepted. Applicability will be determined by the Policy and Standards Division.
F3230–21a	Standard Practice for Safety Assessments of Systems and Equipment in Small Aircraft.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3230–21a is not accepted. Applicability will be determined by the Policy and Standards Division.
F3231/F3231M–23	Standard Specification for Electrical Systems for Aircraft with Combustion Engine Electrical Power Generation.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3231/F3231M–23 is not accepted. Applicability will be determined by the Policy and Standards Division.
F3232/F3232M–20	Standard Specification for Flight Controls in Small Aircraft.	None.	
F3233/F3233M–21	Standard Specification for Flight and Navigation Instrumentation in Aircraft.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3233/F3233M–21 is not accepted. Applicability will be determined by the Policy and Standards Division.
F3234/F3234M–21	Standard Specification for Exterior Lighting in Small Aircraft.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3234/F3234M–21 is not accepted. Applicability will be determined by the Policy and Standards Division.
F3235–22	Standard Specification for Aircraft Storage Batteries.	Remove: Section 4.2	If applying for certification of an airplane with installed lithium batteries, applicants may use the guidance provided by RTCA DO–311A or may obtain FAA acceptance of a different MOC in accordance with § 23.2010.

TABLE 1—PART 23 ACCEPTED MOC BASED ON ASTM CONSENSUS STANDARDS^{4 5}—Continued

ASTM No. as identified in F3264–23	ASTM document title	Changes required for FAA acceptance ⁴	Additional information ⁵
F3236–21a	Standard Specification for High Intensity Radiated Field (HIRF) Protection in Small Aircraft.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3235–22 is not accepted. Applicability will be determined by the Policy and Standards Division.
F3239–22a	Standard Specification for Aircraft Electric Propulsion Systems.	The FAA does not universally accept F3239–22a due to inexperience with the standard.	Aircraft Type Code compliance matrix table found in F3236–21a is not accepted. Applicability will be determined by the Policy and Standards Division.
F3254–22	Standard Specification for Aircraft Interaction of Systems and Structures.	Figures 2, 3, and 4 Replace: “Remote” with: “10 ^{–5} ”. Replace: “Extremely Improbable” with: “10 ^{–8} ” for Level 1, 2 and 3 airplanes and with “10 ^{–9} ” for Level 4 airplanes”.	Applicants are encouraged to consider proposing F3239–22a for development of their MOC for electric propulsion systems on a project-by-project basis. Any MOC proposed must establish a level of safety equivalent to certified reciprocating and turbine propulsion systems and receive acceptance by the FAA in accordance with §23.2010.
F3309/F3309M–21	Standard Practice for Simplified Safety Assessment of Systems and Equipment in Small Aircraft.	None.	Other proposed probabilities will be considered by the FAA on a case-by-case basis.
F3316/F3316M–19	Standard Specification for Electrical Systems for Aircraft with Electric or Hybrid-Electric Propulsion.	FAA does not universally accept F3316/F3316M–19 due to inexperience with the standard.	Applicants are encouraged to consider proposing F3316/F3316M–19 for development of their MOC for electrical systems installed on airplanes with electric or hybrid-electric propulsion systems on a project-by-project basis. Applicants may obtain FAA acceptance of a different MOC in accordance with §23.2010.
F3331–18	Standard Practice for Aircraft Water Loads.	Remove: Table 1	Aircraft Type Code compliance matrix table found in F3316/F3316M–19 is not accepted. Applicability will be determined by the Policy and Standards Division.
F3367–21a	Standard Practice for Simplified Methods for Addressing High-Intensity Radiated Fields (HIRF) and Indirect Effects of Lightning on Aircraft.	None.	Replace: paragraph 5.1.1 with: “Systems that are part of the Type Certificated Engine must be installed in accordance with the engine manufacturer’s requirements. The minimum HIRF and lightning qualification in accordance with sections 8 and 9 of this ASTM practice should be met at the aircraft level, except for engine control systems in Level 1 and 2 airplanes which should meet the following: HIRF: DO–160, section 20–R for both radiated and conducted susceptibility. Lightning: Utilize guidance in AC 33.28–3. For metallic fuselage DO–160G, section 22–A3J3L3 (shielded) and A3H3L3 (unshielded). For composite fuselage DO–160G, section 22–B3K3L3 (shielded) and B3H3L3 (unshielded). Use of lower HIRF and lightning induced voltage & current levels may be acceptable for electronic engine control systems if substantiated at the airplane level (by test in the proposed installation or similar) when exposed to external HIRF environment per AC 20–158 (latest revision) and lightning per AC20–136 (latest revision); using shielding and grounding of the electronic engine control system and accessories in the given installation.”
F3380–19	Standard Practice for Structural Compliance of Very Light Aeroplanes.	None.	
F3396/F3396M–20	Standard Practice for Aircraft Simplified Loads Criteria.	None.	
F3397/F3397M–21	Standard Practice for Aeroplane Turbine Fuel System Hot Weather Operations.	None.	
F3408/F3408M–21	Standard Specification for Aircraft Emergency Parachute Recovery Systems.	None.	
F3432–20a	Standard Practice for Powerplant Instruments.	None.	
F3498–21	Standard Practice for Developing Simplified Fatigue Load Spectra.	None.	

TABLE 1—PART 23 ACCEPTED MOC BASED ON ASTM CONSENSUS STANDARDS^{4 5}—Continued

ASTM No. as identified in F3264–23	ASTM document title	Changes required for FAA acceptance ⁴	Additional information ⁵
F3532–22	Standard Practice for Protection of Aircraft Systems from Intentional Unauthorized Electronic Interactions.	None.	

TABLE 2—SIDE-BY-SIDE VIEW OF PART 23 REGULATIONS AND ASTM F3264–23 SECTION(S)

Part 23 amendments 23–64 and 23–65 regulation(s)	ASTM F3264–23 section(s)1A ⁶	ASTM F3264–23 subsection(s)1A ⁷
Subpart B—Flight		
§ 23.2100 Weight and center of gravity ..	5.1 Weight/Mass and Center of Gravity	5.1.1 F3082/F3082M–22 Standard Specification for Weights and Centers of Gravity of Aircraft 5.1.2 F3114–21 Standard Specification for Structures.
§ 23.2105 Performance data	5.2 Performance Data	5.2.1 F3179/F3179M–22e1 Standard Specification for Performance of Aircraft.
§ 23.2110 Stall speed	5.3 Stall Speed	5.3.1 F3179/F3179M–22e1 Standard Specification for Performance of Aircraft.
§ 23.2115 Takeoff performance	5.4 Takeoff Performance	5.4.1 F3179/F3179M–22e1 Standard Specification for Performance of Aircraft.
§ 23.2120 Climb requirements	5.5 Climb Requirements	5.5.1 F3179/F3179M–22e1 Standard Specification for Performance of Aircraft.
§ 23.2125 Climb information	5.6 Climb Information	5.6.1 F3179/F3179M–22e1 Standard Specification for Performance of Aircraft.
§ 23.2130 Landing	5.7 Landing	5.7.1 F3179/F3179M–22e1 Standard Specification for Performance of Aircraft.
§ 23.2135 Controllability	5.8 Controllability	5.8.1 F3173/F3173M–21a Standard Specification for Aircraft Handling Characteristics.
§ 23.2140 Trim	5.9 Trim	5.9.1 F3173/F3173M–21a Standard Specification for Aircraft Handling Characteristics.
§ 23.2145 Stability	5.10 Stability	5.10.1 F3173/F3173M–21a Standard Specification for Aircraft Handling Characteristics.
§ 23.2150 Stall characteristics, stall warning, and spins.	5.11 Stall Characteristics, Stall Warning, and Spins.	5.11.1 F3180/F3180M–21 Standard Specification for Low-Speed Flight Characteristics of Aircraft.
§ 23.2155 Ground and water handling characteristics.	5.12 Ground and Water Handling Characteristics.	5.12.1 F3173/F3173M–21a Standard Specification for Aircraft Handling Characteristics.
§ 23.2160 Vibration, buffeting, and high-speed characteristics.	5.13 Vibration, Buffeting, and High-Speed Characteristics.	5.13.1 F3173/F3173M–21a Standard Specification for Aircraft Handling Characteristics.
§ 23.2165 Performance and flight characteristics requirements for flight in icing conditions.	5.14 Performance and Flight Characteristics Requirements for Flight in Icing Conditions.	5.14.1 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft.
Subpart C—Structures		
§ 23.2200 Structural design envelope	6.1 Structural Design Envelope	6.1.1 F3116/F3116M–18e2 Standard Specification for Design Loads and Conditions 6.1.1.1 F3396/F3396M–20 Standard Practice for Aircraft Simplified Loads.
§ 23.2205 Interaction of systems and structures.	6.2 Interaction of Systems and Structure.	6.2.1 F3254–22 Standard Specification for Aircraft Interaction of Systems and Structures.
§ 23.2210 Structural design loads	6.3 Structural Design Loads	6.3.1 F3116/F3116M–18e2 Standard Specification for Design Loads and Conditions 6.3.1.1 F3396/F3396M–20 Standard Practice for Aircraft Simplified Loads 6.3.2 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2215 Flight load conditions	6.4 Flight Load Conditions	6.4.1 F3116/F3116M–18e2 Standard Specification for Design Loads and Conditions 6.4.1.1 F3396/F3396M–20 Standard Practice for Aircraft Simplified Loads.
§ 23.2220 Ground and water load conditions.	6.5 Ground and Water Load Conditions	6.5.1 F3116/F3116M–18e2 Standard Specification for Design Loads and Conditions 6.5.1.1 F3331–18 Standard Practice for Aircraft Water Loads.
§ 23.2225 Component loading conditions	6.6 Component Loading Conditions	6.6.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 6.6.1.1 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft 6.6.2 F3116/F3116M–18e2 Standard Specification for Design Loads and Conditions. 6.6.2.1 F3396/F3396M–20 Standard Practice for Aircraft Simplified Loads.
§ 23.2230 Limit and ultimate loads	6.7 Limit and Ultimate Loads	6.7.1 F3114–21 Standard Specification for Structures. 6.7.2 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.

⁴ The MOC are intended for traditional part 23 airplanes, not for novel designs. Novel designs require evaluation and possible modification of the MOC.

⁵ You may find additional information on the FAA Small Airplane Issues List (SAIL) here: https://www.faa.gov/aircraft/air_cert/design_approvals/small_airplanes/small_airplanes_regs/.

⁶ The ASTM F3264–23 section(s) provides a means of compliance intended to be used on projects for traditional part 23 airplanes, not for novel designs. Novel designs require evaluation and possible modification of the means of compliance.

⁷ Changes required for FAA acceptance and additional information per table 1 still applies to table 2. The FAA does not accept the Aircraft Type

Code compliance matrix tables included in F3228–21, F3229/F3229M–17, F3230–21a, F3231/F3231M–23, F3233/F3233M–21, F3234/F3234M–21, F3235–22, F3236–21a, and F3316/F3316M–19. The tables defining applicability found in F3061/F3061M–22b sections 4, 10, 13, and 17 are not accepted. Applicability will be determined by the Policy and Standards Division.

TABLE 2—SIDE-BY-SIDE VIEW OF PART 23 REGULATIONS AND ASTM F3264–23 SECTION(S)—Continued

Part 23 amendments 23–64 and 23–65 regulation(s)	ASTM F3264–23 section(s)1A ⁶	ASTM F3264–23 subsection(s)1A ⁷
§ 23.2235 Structural strength	6.8 Structural Strength	6.8.1 F3114–21 Standard Specification for Structures. 6.8.2 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2240 Structural durability	6.9 Structural Durability	6.9.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 6.9.2 F3066/F3066M –18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation 6.9.3 F3115/F3115M–23 Standard Specification for Structural Durability for Small Aeroplanes. 6.9.3.1 F3380 –19 Standard Practice for Structural Compliance of Very Light Aeroplanes 6.9.3.2 F3498–21 Standard Practice for Developing Simplified Fatigue Load Spectra. 6.9.4 F3116/F3116M–18e2 Standard Specification for Design Loads and Conditions.
§ 23.2245 Aeroelasticity	6.10 Aeroelasticity	6.10.1 F3093/F3093M–21 Standard Specification for Aeroelasticity Requirements.
§ 23.2250 Design and construction principles.	6.11 Design and Construction Principles	6.11.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 6.11.1.1 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 6.11.2 F3114–21 Standard Specification for Structures. 6.11.2.1 F3380–19 Standard Practice for Structural Compliance of Very Light Aeroplanes. 6.11.3 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2255 Protection of structure	6.12 Protection of Structure	6.12.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 6.12.1.1 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 6.12.2 F3114–21 Standard Specification for Structures. 6.12.2.1 F3380–19 Standard Practice for Structural Compliance of Very Light Aeroplanes. 6.12.3 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 6.12.4 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2260 Materials and processes	6.13 Materials and Processes	6.13.1 F3114–21 Standard Specification for Structures. 6.13.1.1 F3380–19 Standard Practice for Structural Compliance of Very Light Aeroplanes. 6.13.2 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2265 Special factors of safety	6.14 Special Factors of Safety	6.14.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 6.14.2 F3114–21 Standard Specification for Structures. 6.14.2.1 F3380 –19 Standard Practice for Structural Compliance of Very Light Aeroplanes.
§ 23.2270 Emergency conditions	6.15 Emergency Conditions	6.15.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 6.15.1.1 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 6.15.2 F3083/F3083M–20a Standard Specification for Emergency Conditions, Occupant Safety and Accommodations. 6.15.3 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.

Subpart D—Design and Construction

§ 23.2300 Flight control systems	7.1 Flight Control Systems	7.1.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 7.1.1.1 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 7.1.2 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 7.1.3 F3117/F3117M–23a Standard Specification for Crew Interface.
§ 23.2305 Landing gear systems	7.2 Landing Gear Systems	7.2.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft.
§ 23.2310 Buoyancy for seaplanes and amphibians.	7.3 Buoyancy for Seaplanes and Amphibians.	7.3.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft.
§ 23.2315 Means of egress and emergency exits.	7.4 Means of Egress and Emergency Exits.	7.4.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 7.4.2 F3083/F3083M–20a Standard Specification for Emergency Conditions, Occupant Safety and Accommodations.

TABLE 2—SIDE-BY-SIDE VIEW OF PART 23 REGULATIONS AND ASTM F3264–23 SECTION(S)—Continued

Part 23 amendments 23–64 and 23–65 regulation(s)	ASTM F3264–23 section(s)1A ⁶	ASTM F3264–23 subsection(s)1A ⁷
§ 23.2320 Occupant physical environment.	7.5 Occupant Physical Environment	7.5.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 7.5.1.1 F3227/F3227M–22 Standard Specification for Environmental Systems in Small Aircraft. 7.5.2 F3083/F3083M–20a Standard Specification for Emergency Conditions, Occupant Safety and Accommodations 7.5.3 F3114–21 Standard Specification for Structures. 7.5.4 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft.
§ 23.2325 Fire protection	7.6 Fire Protection	7.6.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft 7.6.1.1 F3231/F3231M–23 Standard Specification for Electrical Systems for Aircraft with Combustion Engine Electrical Power Generation. 7.6.1.2 F3234/F3234M–21 Standard Specification for Exterior Lighting in Small Aircraft. 7.6.1.3 F3316/F3316M–19 Standard Specification for Electrical Systems for Aircraft with Electric or Hybrid-Electric Propulsion. 7.6.2 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 7.6.3 F3083/F3083M–20a Standard Specification for Emergency Conditions, Occupant Safety and Accommodations. 7.6.4 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2330 Fire protection in designated fire zones and adjacent areas.	7.7 Fire Protection in Designated Fire Zones and Adjacent Areas.	7.7.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 7.7.1.1 F3231/F3231M–23 Standard Specification for Electrical Systems for Aircraft with Combustion Engine Electrical Power Generation. 7.7.2 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 7.7.3 F3114–21 Standard Specification for Structures.
§ 23.2335 Lightning protection	7.8 Lightning Protection	7.8.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft.

Subpart E—Powerplant

§ 23.2400 Powerplant installation	8.1 Powerplant Installation	8.1.1 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 8.1.2 F3063/F3063M–21 Standard Specification for Aircraft Fuel and Energy Storage and Delivery. 8.1.3 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 8.1.3.1 F3432–20a Standard Practice for Powerplant Instruments. 8.1.4 F3065/F3065M–21a Standard Specification for Aircraft Propeller System Installation. 8.1.5 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 8.1.6 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems.
§ 23.2405 Automatic power or thrust control systems.	8.2 Power or Thrust Control Systems ...	8.2.1 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 8.2.2 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 8.2.2.1 F3432–20a Standard Practice for Powerplant Instruments. 8.2.3 F3065/F3065M–21a Standard Specification for Aircraft Propeller System Installation.
§ 23.2410 Powerplant installation hazard assessment.	8.3 Powerplant Installation Hazard Assessment.	8.2.4 F3117/F3117M–23a Standard Specification for Crew Interface. 8.3.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 8.3.2 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation 8.3.3 F3063/F3063M–21 Standard Specification for Aircraft Fuel and Energy Storage and Delivery. 8.3.4 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 8.3.4.1 F3432–20a Standard Practice for Powerplant Instruments. 8.3.5 F3065/F3065M–21a Standard Specification for Aircraft Propeller System Installation. 8.3.6 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 8.3.7 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 8.3.8 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems.
§ 23.2415 Powerplant ice protection	8.4 Powerplant Installation Ice Protection.	8.4.1 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 8.4.2 F3063/F3063M–21 Standard Specification for Aircraft Fuel and Energy Storage and Delivery. 8.4.3 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 8.4.4 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems.

TABLE 2—SIDE-BY-SIDE VIEW OF PART 23 REGULATIONS AND ASTM F3264–23 SECTION(S)—Continued

Part 23 amendments 23–64 and 23–65 regulation(s)	ASTM F3264–23 section(s)1A ⁶	ASTM F3264–23 subsection(s)1A ⁷
§ 23.2420 Reversing systems	8.5 Reversing Systems	8.5.1 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 8.5.2 F3065/F3065M–21a Standard Specification for Aircraft Propeller System Installation. 8.5.3 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems
§ 23.2425 Powerplant operational characteristics.	8.6 Powerplant Operational Characteristics.	8.6.1 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 8.6.2 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 8.6.2.1 F3432–20a Standard Practice for Powerplant Instruments. 8.6.3 F3065/F3065M–21a Standard Specification for Aircraft Propeller System Installation 8.6.4 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 8.6.5 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft 8.6.6 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems.
§ 23.2430 Fuel systems	8.7 Fuel and Energy Storage and Distribution Systems.	8.7.1 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 8.7.2 F3063/F3063M–21 Standard Specification for Aircraft Fuel and Energy Storage and Delivery. 8.7.2.1 F3397/F3397M–21 Standard Practice for Aeroplane Turbine Fuel System Hot Weather Operations 8.7.3 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 8.7.3.1 F3432–20a Standard Practice for Powerplant Instruments. 8.7.4 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 8.7.5 F3114–21 Standard Specification for Structures. 8.7.6 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems.
§ 23.2435 Powerplant induction and exhaust systems.	8.8 Powerplant Induction, Exhaust, and Support Systems.	8.8.1 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 8.8.2 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems.
§ 23.2440 Powerplant fire protection	8.9 Powerplant Installation Fire Protection.	8.9.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft 8.9.2 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation 8.9.3 F3063/F3063M–21 Standard Specification for Aircraft Fuel an Energy Storage and Delivery. 8.9.4 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 8.9.4.1 F3432–20a Standard Practice for Powerplant Instruments. 8.9.5 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 8.9.6 F3239–22a Standard Specification for Aircraft Electric Propulsion Systems.

Subpart F—Equipment

§ 23.2500 Airplane level systems requirements.	9.1 Systems and Equipment Function—Requirements.	9.1.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.1.1.1 F3231/F3231M–23 Standard Specification for Electrical Systems for Aircraft with Combustion Engine Electrical Power Generation. 9.1.1.1(a) F3235–22 Standard Specification for Aircraft Storage Batteries. 9.1.1.2 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 9.1.1.3 F3233/F3233M–21 Standard Specification for Flight and Navigation Instrumentation in Aircraft. 9.1.1.3(a) F3229/F3229M–17 Standard Practice for Static Pressure System Tests in Small Aircraft. 9.1.1.4 F3316/F3316M–19 Standard Specification for Electrical Systems for Aircraft with Electric or Hybrid-Electric Propulsion. 9.1.2 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 9.1.2.1 F3432–20a Standard Practice for Powerplant Instruments. 9.1.3 F3066/F3066M–18 Standard Specification for Aircraft Powerplant Installation Hazard Mitigation. 9.1.4 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 9.1.5 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft. 9.1.6 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
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TABLE 2—SIDE-BY-SIDE VIEW OF PART 23 REGULATIONS AND ASTM F3264–23 SECTION(S)—Continued

Part 23 amendments 23–64 and 23–65 regulation(s)	ASTM F3264–23 section(s)1A ⁶	ASTM F3264–23 subsection(s)1A ⁷
§ 23.2505 Function and installation	9.2 Equipment Function and Installation Requirements.	9.2.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.2.1.1 F3231/F3231M–23 Standard Specification for Electrical Systems for Aircraft with Combustion Engine Electrical Power Generation. 9.2.1.1(a) F3235–22 Standard Specification for Aircraft Storage Batteries. 9.2.1.2 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 9.2.1.3 F3233/F3233M–21 Standard Specification for Flight and Navigation Instrumentation in Aircraft. 9.2.1.4 F3316/F3316M–19 Standard Specification for Electrical Systems for Aircraft with Electric or Hybrid-Electric Propulsion. 9.2.2 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2510 Equipment, systems, and installations.	9.3 Equipment, Systems, and Installation.	9.3.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.3.1.1 F3230–21a Standard Practice for Safety Assessments of Systems and Equipment in Small Aircraft. 9.3.1.2 F3233/F3233M–21 Standard Specification for Flight and Navigation Instrumentation in Aircraft. 9.3.1.3 F3227/F3227M–22 Standard Specification for Environmental Systems in Small Aircraft. 9.3.1.4 F3309/F3309M–21 Standard Practice for Simplified Safety Assessment of Systems and Equipment in Small Aircraft. 9.3.1.5 F3532–22 Standard Practice for Protection of Aircraft Systems from Intentional Unauthorized Electronic Interactions. 9.3.2 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2515 Electrical and electronic system lightning protection.	9.4 Electrical and Electronic System Lightning Protection.	9.4.1 F3061/F3061M–22b22b2322b Standard Specification for Systems and Equipment in Small Aircraft. 9.4.1.1 F3367–21a Standard Practice for Simplified Methods for Addressing High-Intensity Radiated Fields (HIRF) and Indirect Effects of Lightning on Aircraft.
§ 23.2520 High-intensity Radiated Fields (HIRF) protection.	9.5 High Intensity Radiated Fields (HIRF) Protection.	9.5.1 F3061/F3061M–22b22b2322b Standard Specification for Systems and Equipment in Small Aircraft. 9.5.1.1 F3236–21a Standard Specification for High Intensity Radiated Field (HIRF) Protection in Small Aircraft. 9.5.1.2 F3367–21a Standard Practice for Simplified Methods for Addressing High-Intensity Radiated Fields (HIRF) and Indirect Effects of Lightning on Aircraft.
§ 23.2525 System power generation, storage, and distribution.	9.6 System Power Generation, Storage, and Distribution.	9.6.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.6.1.1 F3231/F3231M–23 Standard Specification for Electrical Systems for Aircraft with Combustion Engine Electrical Power Generation. 9.6.1.1(a) F2490–20 Standard Guide for Aircraft Electrical Load and Power Source Capacity Analysis. 9.6.1.2 F3233/F3233M–21 Standard Specification for Flight and Navigation Instrumentation in Aircraft. 9.6.1.3 F3316/F3316M–19 Standard Specification for Electrical Systems for Aircraft with Electric or Hybrid-Electric Propulsion. 9.6.1.3(a) F2490–20 Standard Guide for Aircraft Electrical Load and Power Source Capacity Analysis. 9.6.2 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 9.6.3 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft.
§ 23.2530 External and cockpit lighting ..	9.7 External and Cockpit Lighting	9.7.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.7.1.1 F3233/F3233M–21 Standard Specification for Flight and Navigation Instrumentation in Aircraft. 9.7.1.2 F3234/F3234M–21 Standard Specification for Exterior Lighting in Small Aircraft. 9.7.2 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 9.7.3 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft.
§ 23.2535 Safety equipment	9.8 Safety Equipment	9.8.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.8.2 F3083/F3083M–20a Standard Specification for Emergency Conditions, Occupant Safety and Accommodations. 9.8.3 F3117/F3117M–23a Standard Specification for Crew Interface.
§ 23.2540 Flight in icing conditions	9.9 Flight in Icing Conditions	9.9.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.9.1.1 F3233/F3233M–21 Standard Specification for Flight and Navigation Instrumentation in Aircraft. 9.9.2 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft.
§ 23.2545 Pressurized systems elements.	9.10 Pressurized System Elements	9.10.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 9.10.1.1 F3229/F3229M–17 Standard Practice for Static Pressure System Tests in Small Aircraft.

TABLE 2—SIDE-BY-SIDE VIEW OF PART 23 REGULATIONS AND ASTM F3264–23 SECTION(S)—Continued

Part 23 amendments 23–64 and 23–65 regulation(s)	ASTM F3264–23 section(s)1A ⁶	ASTM F3264–23 subsection(s)1A ⁷
§ 23.2550 Equipment containing high-energy rotors.	9.11 Equipment Containing High-Energy Rotors.	9.11.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft.
Subpart G—Flight crew interface and Other Information		
§ 23.2600 Flightcrew interface	10.1 Flight crew Compartment Interface	10.1.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 10.1.1.1 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 10.1.2 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 10.1.3 F3063/F3063M–21 Standard Specification for Aircraft Fuel and Energy Storage and Delivery. 10.1.4 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 10.1.4.1 F3432–20a Standard Practice for Powerplant Instruments. 10.1.5 F3114–21 Standard Specification for Structures. 10.1.6 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 10.1.7 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2605 Installation and operation	10.2 Installation and Operation Information.	10.2.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 10.2.1.1 F3227/F3227M–22 Standard Specification for Environmental Systems in Small Aircraft. 10.2.1.2 F3231/F3231M–23 Standard Specification for Electrical Systems for Aircraft with Combustion Engine Electrical Power Generation. 10.2.1.3 F3232/F3232M–20 Standard Specification for Flight Controls in Small Aircraft. 10.2.1.4 F3233/F3233M–21 Standard Specification for Flight and Navigation Instrumentation in Aircraft. 10.2.2 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation. 10.2.3 F3063/F3063M–21 Standard Specification for Aircraft Fuel and Energy Storage and Delivery. 10.2.4 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 10.2.4.1 F3432–20a Standard Practice for Powerplant Instruments. 10.2.5 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 10.2.6 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft. 10.2.7 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2610 Instrument markings, control markings, and placards.	10.3 Instrument Markings, Control Markings, and Placards.	10.3.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 10.3.2 F3063/F3063M–21 Standard Specification for Aircraft Fuel and Energy Storage and Delivery. 10.3.3 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 10.3.4 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft. 10.3.5 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
§ 23.2615 Flight, navigation, and powerplant instruments.	10.4 Flight, Navigation, and Powerplant Instruments.	10.4.1 F3061/F3061M–22b Standard Specification for Systems and Equipment in Small Aircraft. 10.4.2 F3062/F3062M–20 Standard Specification for Aircraft Powerplant Installation 10.4.3 F3064/F3064M–21 Standard Specification for Aircraft Powerplant Control, Operation, and Indication. 10.4.3.1 F3432–20a Standard Practice for Powerplant Instruments. 10.4.4 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft.
§ 23.2620 Airplane flight manual	5.15 Operating Limitations	5.15.1 F3174/F3174M–21 Standard Specification for Establishing Operating Limitations and Information for Aeroplanes. 5.15.2 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.
	10.5 Airplane Flight Manual	10.5.1 F3117/F3117M–23a Standard Specification for Crew Interface in Aircraft. 10.5.2 F3174/F3174M–21 Standard Specification for Establishing Operating Limitations and Information for Aeroplanes. 10.5.3 F3120/F3120M–20 Standard Specification for Ice Protection for General Aviation Aircraft. 10.5.4 F3408/F3408M–21 Standard Specification for Aircraft Emergency Parachute Recovery Systems.

The MOC accepted by this NOA provide one means, but not the only means of complying with part 23 regulatory requirements. Applicants who desire to use an MOC reflected by

other revisions to ASTM standards not previously accepted may seek guidance and possible acceptance from the FAA for the use of those MOC on a case-by-case basis. Applicants may propose an

alternative MOC for FAA review and possible acceptance.

Editorial, Reapproval, Revision, or Withdrawal

ASTM policy is that a consensus standard should be reviewed in its entirety by the responsible subcommittee and must be balloted for reapproval, revision, or withdrawal, within five years of its last approval date. When an ASTM standard is *reapproved*, that reapproval is denoted by the year in parentheses (e.g., F2427–05a (2013)). This date indicates the completion of a review cycle with no technical changes made to the standard. In addition, ASTM issues *editorial changes* denoted by a superscript epsilon in the standard designation (e.g., F3235–17^{ε1}). This epsilon indicates information was corrected, and it did not change the meaning or intent of a standard. Since *reapprovals* and *editorial changes* do not change the technical content of standards, then any standard FAA-accepted by this NOA that is later *reapproved* or *editorially changed* by ASTM, is also considered FAA-accepted and without the need for an updated NOA.

ASTM *revises* a standard to make changes to its technical content. Revisions are identified by a hyphen after the document number, which is followed by the last two numbers of the year of acceptance or of last revision. If the standard is revised again during the same year, this is indicated by adding an “a” for the second revision, “b” for the third revision, etc. Since *revisions* change the technical content, *revisions* to consensus standards serving as the basis for an MOC accepted by this NOA will not be automatically accepted and will require further FAA acceptance for the revisions to be an accepted MOC.

Availability

ASTM F3264–23, “Standard Specification for Normal Category Aeroplanes Certification,” is available online at <https://www.astm.org/READINGLIBRARY/>. ASTM copyrights these consensus standards and charges the public a fee for service. Individual downloads or reprints of a standard (single or multiple copies, or special compilations, and other related technical information) may be obtained online or by contacting ASTM by telephone: (610) 832–9585; facsimile: (610) 832–9555; or through email: service@astm.org. To inquire about consensus standard content and/or membership or about ASTM Offices abroad, contact Joe Koury, Staff Manager for Committee F44 on General Aviation Aircraft by telephone: (610) 832–9804 or through email: jkoury@astm.org.

Issued in Kansas City, Missouri, on May 21, 2024.

Patrick Mullen,

Manager, Technical Policy Branch, Policy and Standards Division, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2024–0221; Project Identifier AD–2023–01233–T; Amendment 39–22762; AD 2024–11–01]

RIN 2120–AA64

Airworthiness Directives; Airbus SAS Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2021–21–02, which applied to certain Airbus SAS Model A318, A319, A320, A321, A330–200, A330–200 Freighter, A330–300, A330–800, A330–900, A340–200, A340–300, A340–500, A340–600, and A380–800 series airplanes. AD 2021–21–02 required replacing certain parts manufacturer approval (PMA) Ni-Cd batteries with serviceable Ni-Cd batteries or maintaining the electrical storage capacity of those PMA Ni-Cd batteries during airplane storage or parking. This AD was prompted by a determination that the on-wing preservation procedures originally provided in that AD did not ensure the expected preservation of the battery capacity. This AD adds airplanes to the applicability and requires replacing each affected part with a serviceable part before release to service of an airplane after a storage or parking period, as applicable. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective July 3, 2024.

ADDRESSES:

AD Docket: You may examine the AD docket at [regulations.gov](https://www.regulations.gov) under Docket No. FAA–2024–0221; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200

New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 206–231–3225; email dan.rodina@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2021–21–02, Amendment 39–21762 (86 FR 62898, November 15, 2021) (AD 2021–21–02). AD 2021–21–02 applied to certain Airbus SAS Model A318, A319, A320, A321, A330–200, A330–200 Freighter, A330–300, A330–800, A330–900, A340–200, A340–300, A340–500, A340–600, and A380–800 series airplanes. The NPRM published in the **Federal Register** on February 9, 2024 (89 FR 9074). The NPRM was prompted by a determination that the on-wing preservation procedures originally provided in that AD did not ensure the expected preservation of the battery capacity. In the NPRM, the FAA proposed to add Model A300 series airplanes; Model A300 B4–600, B4–600R, and F4–600R series airplanes, and Model A300 C4–605R Variant F airplanes (collectively called Model A300–600 series airplanes); Model A310 series airplanes; and Model A350–941 and –1041 airplanes to the applicability. The FAA proposed that the superseding AD would retain none of the requirements of AD 2021–21–02. The FAA proposed to require replacing each affected part with a serviceable part before release to service of an airplane after a storage or parking period, as applicable. The FAA is issuing this AD to address reduced battery endurance performance, which could possibly result in failure to supply the minimum essential electrical power during abnormal or emergency conditions.

Discussion of Final Airworthiness Directive

Comments

The FAA received comments from Air Line Pilots Association, International (ALPA), who supported the SNPRM/NPRM without change.

The FAA received additional comments from American Airlines (AA) and United Airlines (UAL). The following presents the comments received on the NPRM and the FAA’s response to each comment.

Request for Clarification of Terms

AAL requested clarification of “parking and storage” as intended by