## **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. FAA-2024-1702; Project Identifier MCAI-2024-00067-T]

#### RIN 2120-AA64

# Airworthiness Directives; Airbus SAS Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking

(NPRM).

**SUMMARY:** The FAA proposes to supersede Airworthiness Directive (AD) 2021-25-14, which applies to all Airbus SAS Model A319-111, -112, -113, -114, -115, -131, -132, and -133 airplanes; Model A320-211, -212, -214, -216, -231, -232, and -233 airplanes; and Model A321–111, –112, –131, –211, -212, -213, -231, and -232 airplanes. AD 2021-25-14 requires repetitive inspections for cracking at the wing manhole access panel attachment holes at certain wing skin panels, and corrective action if necessary. Since the FAA issued AD 2021–25–14, new investigation results identified that the applicability of these inspections must be expanded to all airplanes in an affected configuration, and the associated compliance time must be adapted to these respective configurations. In addition, further investigation results found that the postrepair inspection program tasks in accordance with the Structural Repair Manual for the affected area were inadequate. This proposed AD would continue to require the actions in AD 2021-25-14 and would require expanding the applicability to include New Engine Option (NEO) airplanes and accomplishment of the required actions within updated compliance times, as applicable to airplane configuration, as specified in a European Union Aviation Safety Agency (EASA) AD, which is proposed for incorporation by reference (IBR). The FAA is proposing this AD to address the unsafe condition on these products.

**DATES:** The FAA must receive comments on this proposed AD by August 19, 2024.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- Federal eRulemaking Portal: Go to regulations.gov. Follow the instructions for submitting comments.
  - Fax: 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.
- Hand Delivery: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

AD Docket: You may examine the AD docket at regulations.gov under Docket No. FAA–2024–1702; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The street address for Docket Operations is listed above.

Material Incorporated by Reference:

- For the EASA ADs identified in this NPRM, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; website easa.europa.eu. You may find this material on the EASA website at ad.easa.europa.eu. It is also available at regulations.gov under Docket No. FAA–2024–1702.
- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

## FOR FURTHER INFORMATION CONTACT:

Timothy Dowling, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: 817–222–5102; email: Timothy.P.Dowling@faa.gov.

## SUPPLEMENTARY INFORMATION:

## **Comments Invited**

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under ADDRESSES. Include "Docket No. FAA-2024-1702; Project Identifier MCAI-2024-00067-T" at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this proposal because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to regulations.gov, including any personal

information you provide. The agency will also post a report summarizing each substantive verbal contact received about this NPRM.

#### **Confidential Business Information**

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this NPRM contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this NPRM, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as "PROPIN." The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this NPRM. Submissions containing CBI should be sent to Timothy Dowling, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: 817-222-5102; email: Timothy.P.Dowling@faa.gov. Any commentary that the FAA receives which is not specifically designated as CBI will be placed in the public docket for this rulemaking.

## **Background**

The FAA issued AD 2021-25-14, Amendment 39-21858 (86 FR 72171, December 21, 2021) (AD 2021-25-14), for all Airbus SAS Model A319–111, -112, -113, -114, -115, -131, -132, and -133 airplanes; Model A320-211, -212, -214, -216, -231, -232, and -233 airplanes; and Model A321-111, -112, -131, -211, -212, -213, -231, and -232 airplanes. AD 2021-25-14 was prompted by an MCAI originated by EASA, which is the Technical Agent for the Member States of the European Union. EASA issued AD 2021-0256, dated November 16, 2021, to correct an unsafe condition.

AD 2021–25–14 requires repetitive inspections at the wing manhole access panel attachment holes at certain wing skin panels on airplanes with Sharklets or its structural reinforcements installed for cracking of the area, and corrective action if necessary. The FAA issued AD 2021–25–14 to address this condition, which could lead to crack propagation, possibly resulting in the reduced structural integrity of the wings.

# Actions Since AD 2021–25–14 Was Issued

Since the FAA issued AD 2021–25–14, EASA superseded EASA AD 2021–

0256 and issued EASA AD 2024-0027, dated January 25, 2024 (EASA AD 2024-0027) (also referred to as the MCAI), to correct an unsafe condition for all Airbus SAS Model A319-111, -112, -113, -114, -115, -131, -132,–133, –151N, –153N, and –171N airplanes; Model A320–211, –212, –214, -215, -216, -231, -232, -233, -251N, -252N, -253N, -271N, -272N, and -273N airplanes; and Model A321-211, -212, -213, -231, -232, -251N, -252N, -253N, -271N, -272N, -251NX, -252NX, -253NX, -271NX, and -272NX airplanes. Model A320–215 airplanes are not certificated by the FAA and are not included on the U.S. type certificate data sheet; this proposed AD therefore does not include those airplanes in the applicability. Additionally, Airbus SAS Model A321-100 series airplanes were inadvertently included in EASA AD 2021–0256, and also in corresponding AD 2021-25-14, despite the Sharklet device not being installed on these airplanes. This AD affects airplanes that have the Sharklet device installed. Therefore, Model A321-100 series airplanes should not have been included in EASA AD 2021-0256 and was corrected with the issuance of EASA AD 2024-0027, and therefore not included in the applicability of this proposed AD. The MCAI states new investigation results highlighted that these inspections must be applied to all models of A319, A320 and A321 airplanes in an affected configuration, and the associated compliance time must be adapted to these configurations. It has been determined that fatigue cracking may occur in affected areas on airplanes having Sharklets installed during production or in service. This condition, if not detected and corrected, could lead to crack initiation and propagation, possibly resulting in reduced structural integrity of the wings.

The FAA is proposing this AD to address the unsafe condition on these products. You may examine the MCAI in the AD docket at *regulations.gov* under Docket No. FAA–2024–1702.

## **Explanation of Retained Requirements**

Although this proposed AD does not explicitly restate the requirements of AD 2021–25–14, this proposed AD would retain all of the requirements of AD 2021–25–14. Those requirements are referenced in EASA AD 2024–0027, which, in turn, is referenced in paragraph (g) of this proposed AD.

# **Related Service Information Under 1 CFR Part 51**

EASA AD 2024–0027 specifies procedures for repetitive detailed visual inspections of the affected areas, and applicable corrective actions. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

### **FAA's Determination**

This product has been approved by the aviation authority of another country and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, it has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA is issuing this NPRM after determining that the unsafe condition described previously is likely to exist or develop in other products of the same type design.

# Proposed AD Requirements in This NPRM

This proposed AD would retain all requirements of AD 2021–25–14. This proposed AD would require accomplishing the actions specified in

EASA AD 2024–0027 described previously, except for any differences identified as exceptions in the regulatory text of this proposed AD.

# **Explanation of Required Compliance Information**

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA developed a process to use some civil aviation authority (CAA) ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has been coordinating this process with manufacturers and CAAs. As a result, the FAA proposes to incorporate EASA AD 2024-0027 by reference in the FAA final rule. This proposed AD would, therefore, require compliance with EASA AD 2024-0027 in its entirety through that incorporation, except for any differences identified as exceptions in the regulatory text of this proposed AD. Using common terms that are the same as the heading of a particular section in EASA AD 2024-0027 does not mean that operators need comply only with that section. For example, where the AD requirement refers to "all required actions and compliance times," compliance with this AD requirement is not limited to the section titled "Required Action(s) and Compliance Time(s)" in EASA AD 2024-0027. Service information required by EASA AD 2024-0027 for compliance will be available at regulations.gov under Docket No. FAA-2024-1702 after the FAA final rule is published.

## **Costs of Compliance**

The FAA estimates that this AD, if adopted as proposed, would affect 1,650 airplanes of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:

## ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
14 work-hours × \$85 per hour = \$1,190	\$0	\$1,190	\$1,963,500 per inspection cycle.

The FAA has received no definitive data on which to base the cost estimates for the on-condition actions specified in this proposed AD.

## Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and

procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

## **Regulatory Findings**

The FAA determined that this proposed AD would not have federalism

implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Would not affect intrastate aviation in Alaska, and
- (3) Would not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

## § 39.13 [Amended]

- 2. The FAA amends § 39.13 by:
- a. Removing Airworthiness Directive (AD) 2021–25–14, Amendment 39–21858 (86 FR 72171, December 21, 2021); and
- b. Adding the following new AD:

Airbus SAS: Docket No. FAA-2024-1702; Project Identifier MCAI-2024-00067-T.

## (a) Comments Due Date

The FAA must receive comments on this airworthiness directive (AD) by August 19, 2024.

### (b) Affected ADs

This AD replaces AD 2021–25–14, Amendment 39–21858 (86 FR 72171, December 21, 2021) (AD 2021–25–14).

#### (c) Applicability

This AD applies to all Airbus SAS airplanes identified in paragraphs (c)(1) through (3) of this AD and certificated in any category.

- (1) Model A319–111, –112, –113, –114, –115, –131, –132, –133, –151N, –153N, and –171N airplanes.
- (2) Model A320–211, –212, –214, –216, –231, –232, –233, –251N, –252N, –253N, –271N, –272N, –273N airplanes.
- (3) Model A321–211, –212, –213, –231, –232, –251N, –251NX, –252N, –252NX, –253N, –253NX, –271N, –271NX, –272N, and –272NX airplanes.

#### (d) Subject

Air Transport Association (ATA) of America Code 57, Wings.

#### (e) Unsafe Condition

This AD was prompted by new investigation results that identified that these inspections must be applied to all models of A319, A320 and A321 airplanes in an affected configuration, and the associated compliance time must be adapted to these configurations. The FAA is issuing this AD to address fatigue cracking that may occur in affected areas on airplanes having Sharklets installed during production or in service. The unsafe condition, if not addressed, could result in crack initiation and propagation, possibly resulting in reduced structural integrity of the wings.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

## (g) Requirements

Except as specified in paragraphs (h) and (i) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2024–0027, dated January 25, 2024 (EASA AD 2024–0027).

## (h) Exceptions to EASA AD 2024-0027

- (1) Where EASA AD 2024–0027 refers to its effective date, this AD requires using the effective date of this AD.
- (2) Where paragraph (2) of EASA AD 2024-0027 specifies "if, during any inspection as required by paragraph (1) of this AD, any discrepancy is detected as defined in the [Airbus Alert Operators Transmission] AOT, before next flight, contact Airbus for approved repair instructions and, within the compliance time specified therein, accomplish those instructions accordingly," this AD requires replacing that text with "if any cracking is detected, the cracking must be repaired before further flight using a method approved by the Manager, International Validation Branch, FAA; or EASA; or Airbus SAS's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.'
- (3) This AD does not adopt paragraph (4) of EASA AD 2024–0027.
- (4) This AD does not adopt the "Remarks" section of EASA AD 2024–0027.

## (i) No Reporting Requirement

Although the service information referenced in EASA AD 2024–0027 specifies to submit certain information [and send removed parts] to the manufacturer, this AD does not include that requirement.

#### (j) Additional AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your

request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the International Validation Branch, mail it to the address identified in paragraph (k) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov.

(i) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office. (ii) AMOCs approved previously for AD 2021–25–14 are approved as AMOCs for the corresponding provisions of EASA AD 2024–0027 that are required by paragraph (g) of this AD.

(2) Contacting the Manufacturer: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Validation Branch, FAA; or EASA; or Airbus SAS's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) Required for Compliance (RC): Except as required by paragraphs (i) and (j)(2) of this AD, if any service information referenced in EASA AD 2024-0027 that contains paragraphs that are labeled as RC, the instructions in RC paragraphs, including subparagraphs under an RC paragraph, must be done to comply with this AD; any paragraphs, including subparagraphs under those paragraphs, that are not identified as RC are recommended. The instructions in paragraphs, including subparagraphs under those paragraphs, not identified as RC may be deviated from using accepted methods in accordance with the operator's maintenance or inspection program without obtaining approval of an AMOC, provided the instructions identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to instructions identified as RC require approval of an AMOC.

## (k) Additional Information

For more information about this AD, contact Timothy Dowling, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: 817–222–5102; email: *Timothy.P.Dowling@faa.gov*.

## (l) Material Incorporated by Reference

- (1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.
- (2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.
- (i) European Union Aviation Safety Agency (EASA) AD 2024–0027, dated January 25, 2024.
  - (ii) [Reserved]
- (3) For EASA AD 2024–0027, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email *ADs*@easa.europa.eu; website easa.europa.eu. You may find these EASA ADs on the EASA website at ad.easa.europa.eu.
- (4) You may view this material at the FAA, Airworthiness Products Section, Operational

Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit www.archives.gov/federal-register/cfr/ibr-locations, or email fr.inspection@nara.gov.

Issued on June 25, 2024.

#### Suzanne Masterson,

Deputy Director, Integrated Certificate Management Division, Aircraft Certification Service.

[FR Doc. 2024–14438 Filed 7–2–24; 8:45 am]

BILLING CODE 4910-13-P

#### **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. FAA-2024-1703; Project Identifier MCAI-2023-01054-T]

#### RIN 2120-AA64

Airworthiness Directives; Airbus Canada Limited Partnership (Type Certificate Previously Held by C Series Aircraft Limited Partnership (CSALP); Bombardier, Inc.) Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for Airbus Canada Limited Partnership Model BD-500-1A11 airplanes. This proposed AD was prompted by a design review of aircraft structural and stress reports that resulted in a revision of operational loads for some aircraft flight phases. This proposed AD would require using a certain version of the aircraft structural repair manual (ASRP) and a review and disposition of repairs based on previous versions, as specified in a Transport Canada AD, which is proposed for incorporation by reference (IBR). The FAA is proposing this AD to address the unsafe condition on these

**DATES:** The FAA must receive comments on this proposed AD by August 19, 2024

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

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Material Incorporated by Reference:

- For material that is proposed for IBR in this AD, contact Transport Canada, Transport Canada National Aircraft Certification, 159 Cleopatra Drive, Nepean, Ontario K1A 0N5, Canada; telephone 888–663–3639; email TC.AirworthinessDirectives-Consignesdenavigabilite.TC@tc.gc.ca. You may find this material on the Transport Canada website at tc.canada.ca/en/aviation. It is also available at regulations.gov under Docket No. FAA–2024–1703.
- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

FOR FURTHER INFORMATION CONTACT: Deep Gaurav, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516– 228–7300; email 9-avs-nyaco-cos@ faa.gov.

## SUPPLEMENTARY INFORMATION:

## **Comments Invited**

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under ADDRESSES. Include "Docket No. FAA-2024-1703; Project Identifier MCAI-2023-01054-T" at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this proposal because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to regulations.gov, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this NPRM.

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## **Background**

Transport Canada, which is the aviation authority for Canada, has issued Transport Canada AD CF-2023-70, dated October 5, 2023 (Transport Canada AD CF-2023-70) (also referred to as the MCAI), to correct an unsafe condition for all Airbus Canada Limited Partnership Model BD-500-1A11 airplanes. The MCAI states that a design review of aircraft structural and stress reports for model BD-500-1A10 and BD-500-1A11 airplanes has resulted in a revision of operational loads for some aircraft flight phases, affecting certain aircraft sections. As a result, repairs and damage assessments accomplished on aircraft to date may have exceeded the available structural margins and require review to ensure they comply with the revised stress data for the affected sections. The MCAI also states that Transport Canada AD CF-2023-37, dated May 30, 2023, mandates that ASRP 136.01 or later approved versions, or Airbus Canada source data approved at the time of the disposition, is to be used for any new structural assessments, repairs and dispositions